HANDBOOK OF INSTRUCTIONS WITH PARTS CATALOG

Models C532S and C5315S

HOLLOW STEEL THREE BLADE PROPELLERS

(CURTISS)

NOTICE: This document contains information affecting the national defense of the United States within the meaning of the Espionage Act, 50 U. S. C., 31 and 32, as amended. Its transmission or the revelation of its contents in any manner to an unauthorized person is prohibited by law.

Published under joint authority of the Commanding General, Army Air Forces, the Chief of the Bureau of Aeronautics, and the Air Council of the United Kingdom.

THIS PUBLICATION MAY BE USED BY PERSONNEL RENDERING SERVICE TO THE UNITED STATES OR ITS

Instructions Applicable to AAF Personnel.

Paragraph 5.d. of Army Regulation 380-5 relative to the handling of restricted printed matter is quoted below:

"d. Dissemination of restricted matter.—The information contained in restricted documents and the essential characteristics of restricted material may be given to any person known to be in the service of the United States and to persons of undoubted loyalty and discretion who are cooperating in Government work, but will not be communicated to the public or to the press except by authorized military public relations agencies."

Instructions Applicable to Navy Personnel.

Navy Regulations, Article 751/2, contains the following paragraphs relating to the handling of restricted matter:

"(b) Restricted matter may be disclosed to persons of discretion in the Government service when it appears to be in the public interest.

"(c) Restricted matter may be disclosed, under special circumstance., to persons not in the Government service when it appears to be in the public interest."

The Bureau of Aeronautics Circular Letter No. 12-43 further states:

"Therefore, it is requested that all paval activities check their own local regulations and procedures to make sure that handbooks, service instructions and other restricted technical publications are actually being made available to both civilian and enlisted personnel who have use for them."

General.

These instructions permit the issue of restricted publications to civilian contract and other accredited schools engaged in training personnel for Government work, to civilian concerns contracting for overhaul and repair of aircraft or aircraft accessories, and to similar commercial organizations.

-LIST OF REVISED PAGES ISSUÉD-

NOTE: A heavy black vertical line, to the left of the text on revised pages, indicates the extent of the revision. This line is omitted where more than 50 percent of the page is revised.

 Page
 Latest

 No.
 Revised Date

 i
 April 5, 1944

 24
 April 5, 1944

PARTS CATALOG

25-69 Inc.April 5, 1944

ADDITIONAL COPIES OF THIS PUBLICATION MAY BE OBTAINED AS FOLLOWS:

AAF ACTIVITIES.—Submit requisitions to the Commanding General, Fairfield Air Service Command, Patterson Field, Fairfield, Ohio, Attention: Publications Distribution Branch, in accordance with AAF Regulation No. 5-9. Also, for details of Technical Order distribution, see T. O. No. 00-25-3.

NAVY ACTIVITIES.—Submit requests to the Chief, Bureau of Aeronautics, Navy Department, Washington, D. C. Also, see NavAer 00-500 for details on distribution of technical publications.

BRITISH ACTIVITIES.—Submit requirements on Form 294A, in duplicate, to the Air Publications and Forms Store, New College, Leadhall Lane, Harrogate, Yorkshire, England.

TABLE OF CONTENTS

	Subject	Page
I	INTRODUCTION	1
II	DESCRIPTION	1
	1. General Description	
	2. Detailed	1
III	INSTALLATION	4
	1. General	
•	2. Installation	4
IV	OPERATION	
	1. Principles of Operation	
	2. Operating Instructions	8
V	SERVICE INSPECTION, MAINTENANCE, AND LUBRICATION	
	1. Service Tools	
	Inspection and Maintenance Lubrication	
VI	DISASSEMBLY, INSPECTION, REPAIR, AND REASSEMBLY	
	2. Disassembly	
	3. Inspection, and Repair	
	4. Reassembly	
VII	TEST PROCEDURE	22
	1. Preload Adjustment	
	2. Adjustment of Angle Range	
	3. Balancing	
	4. Operational Inspection	23
	PARTS CATALOG	
I	INTRODUCTION	25
П	GROUP ASSEMBLY PARTS LIST	27
Ш	NUMERICAL PARTS LIST	
IV	STANDARD PARTS LIST	68
V	SERVICE TOOLS	69

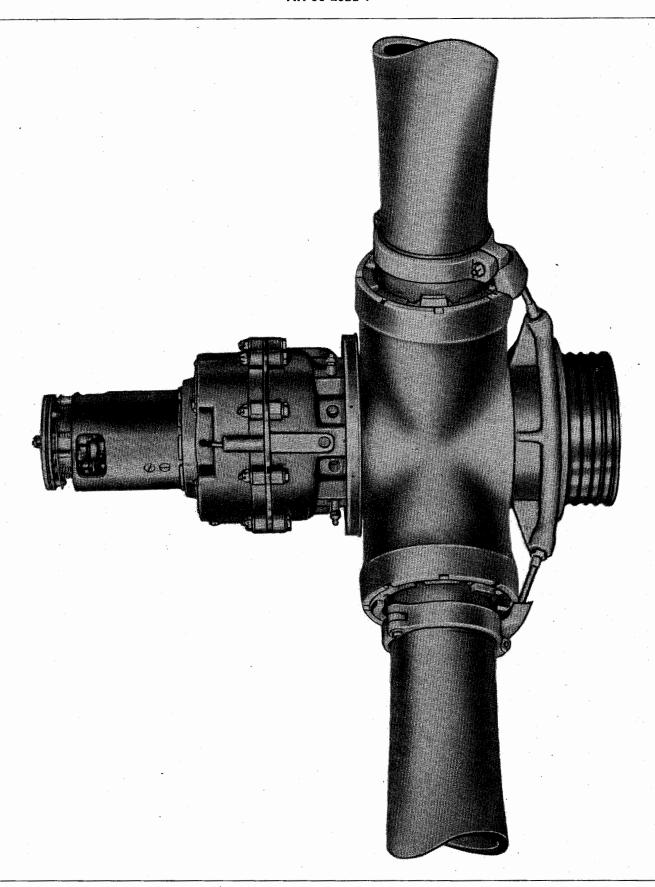


Figure 1—Propeller Assembly
RESTRICTED

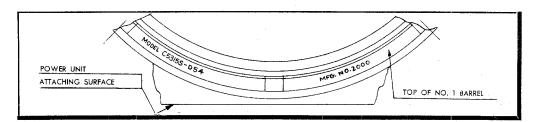
SECTION I

- 1. This Handbook is issued as the basic Technical Instructions for the equipment involved.
- 2. This Handbook contains descriptive data and instructions for the Installation, Operation, Maintenance, Overhaul, and Testing of the Model C5315S (Early

Type) Three Blade Hollow Steel Propeller manufactured by the Curtiss Propeller Division, Curtiss-Wright Corporation, Caldwell, New Jersey, under Contracts W535 ac-14085, ac-16687, ac-16957, ac-17243, and ac-25000.

SECTION II DESCRIPTION

Figure 2—
Propeller Designation



1. GENERAL DESCRIPTION.

The various models of the Electric Propeller are identified by combinations of numbers and letters stamped on the hub of the end of the number one blade barrel.

- a. The first letter "C" signifies Curtiss as the manufacturer.
- b. The first digit of the model number indicates the SAE standard spline shaft upon which the hub will fit. For example, the digit "5" indicates a No. 50 shaft, and "6" indicates a No. 60 shaft size.
 - c. The second digit indicates the number of blades.
- d. The third, or third and fourth digits indicate the blade shank size. For example, the digit "2" indicates a No. 2 blade shank size; "15" indicates a No. 1½ blade shank size.
- e. The letter immediately following the above digits indicates the material from which the blade is fabricated; namely, D for aluminum alloy, S for steel, W for wood.
- f. Any additional letter included with, but after, the material designation letter indicates a special feature of the propeller. Example: C6315SH-C4 indicates a hollow shaft propeller; C5315SP-A2 indicates a pusher type propeller.
- g. A suffix combination, consisting of a letter and one or two digits, indicates the general series to which the propeller belongs and identifies the complete propeller assembly. Early models lacked the suffix letter and digit combination.
- b. The above example indicates a Curtiss Propeller with a No: 50 SAE standard spline, three blades, a No. 1.5 shank, blades of steel, general design series "D,"

specific assembly No. 54 of the preceding designations and Manufacturer's Serial No. 2000.

2. DETAILED DESCRIPTION.

a. HUB ASSEMBLY. (See figure 3.)

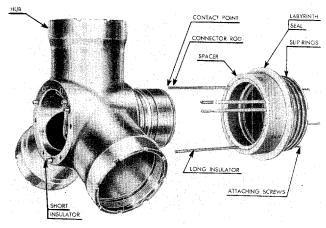


Figure 3—Hub Assembly

- (1) The hub is machined from a single solid forging of alloy steel. On the rear of the hub are mounted four bronze slip rings insulated from the hub and from each other. Four insulated brass connector rods carry the electrical circuits through passages from the slip rings to contacts at the front face of the hub. The blade assemblies are inserted into the hub barrels and are held in place by retaining nuts.
- (2) When specified, anti-icing equipment is included as a component part of the propeller assembly.

b. BLADE ASSEMBLY. (See figure 4.)

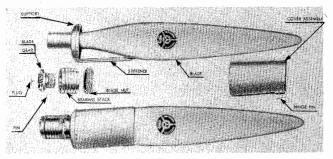


Figure 4—Blade and Cuff Assemblies

- (1) The blade is made by welding together two formed sheets of a steel alloy processed in an electrical furnace. The shank and camber faces are formed from one sheet, and the thrust face from the other. Atomic hydrogen welding, which gives a high-strength and uniform weld, is used to join the two component plates along the leading and trailing edges in a seam that converges at the shank and extends to the root end of the blade. The root end of the blade is internally threaded to receive a spiral bevel gear which is pinned in place. A stack of angular-contact type, antifriction bearings is placed on the blade shank together with a retaining nut. The bearing stack permits free rotation of the blade in the hub under high centrifugal loads.
- (2) Blade shank cuffs, designed to assist the distribution of air to radial engine cylinders, may also be placed on the blade. The cuff consists of a cast magnesium and a stiffener to which a formed sheet (cover assembly) is attached by screws. The entire cuff assembly is held in position by an integral shoulder on the blade.
- (3) Blade markings on the cambered side of the hollow steel propeller blades will be as follows:
- (a) Between the 18 and 24 inch stations the cambered side of each propeller blade will bear markings as described below. The size of the letters and numbers will be $\frac{1}{2}$ inch.
 - 1. Blade serial number.
 - 2. Drawing number (Part No.).
 - 3. Low blade angle at reference station.
 - 4. High blade angle at reference station.

NOTE

The foregoing data will be painted or stenciled with black enamel. When the blade is

- camouflaged the data will be applied with yellow lacquer the same as applied to the tip section. The markings will be protected by a coat of clear lacquer or spar varnish. In no instance will such markings be indented or cut into the metal.
- c. POWER GEAR ASSEMBLY.—The power gear is a bevel gear which meshes with the blade gears and is internally splined to engage with the low speed splined drive of the speed reducer. This power gear is equipped with an angular-contact type thrust bearing which absorbs the power gear thrust and is mounted in a steel adapter plate. The adapter plate also serves as a mounting for the power unit cover or propeller spinner, if such is used. (See figure 5.)

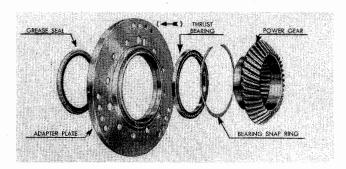


Figure 5—Power Gear Assembly

- d. SPEED REDUCER ASSEMBLY.—The speed reducer consists of two stages of planetary-type reduction gearing contained within an aluminum-alloy housing. The rotating parts are fitted with ball bearings to provide maximum efficiency and also to facilitate assembly. The gear teeth are surface hardened to insure long service life. Gaskets between the front and rear housings and seals at each end make the unit oil-tight and eliminate the necessity for frequent lubrication of the enclosed parts. The unit is partially filled with an oil having an extremely low pour-point, thus providing a continuous oil bath for the speed reducer gears. The low pour-point oil insures unrestricted operation at the low operating temperatures encountered at high altitudes.
- e. LIMIT SWITCH ASSEMBLIES.—At the hub end of the speed reducer are located blade angle limit switches which are operated by pivot arms riding on a cam attached to the low speed bell gear of the speed reducer. The limit switches, which are connected in the

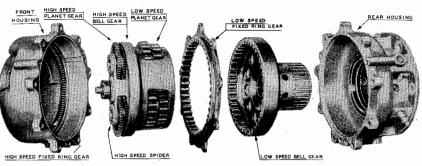


Figure 6—Speed Reducer
Assembly

electric motor leads, are spring-loaded electrical contacts which, upon installation of the power unit, mate with the fixed contacts on the front face of the hub. As the switch arms ride on their respective cam lobes, the contacts are retracted and the circuits are opened. By their location, the cam lobes accurately control the LOW, HIGH, and FEATHER blade angle settings. Reverse thrust propellers have an additional limit switch to halt the pitch change at the NEGATIVE angle setting.

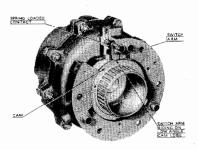


Figure 7— Limit Switch Assembly

f. MAGNETIC BRAKE ASSEMBLY. — The diaphragm type brake consists of a brake disc assembly keyed to the armature shaft, and a diaphragm assembly held against the brake disc by coil springs. The brake solenoid coil is located behind the diaphragm assembly and is connected in series with the motor. When the motor is operated, the solenoid is energized, thereby releasing the brake. When the motor is not operating, the solenoid is not energized; therefore, the brake is applied by the spring forces.

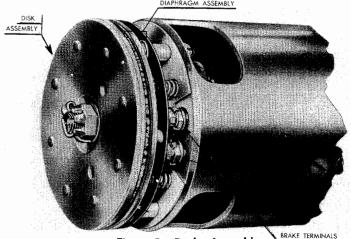


Figure 8—Brake Assembly

g. ELECTRIC MOTOR ASSEMBLY. — The pitch change motor is attached to the front housing of the speed reducer and the armature is fitted to the driving pinion of the high-speed stage of the speed reducer unit. The motor is of the series type and has two field windings which provide for rotation in either direction.

b. BRUSH AND HOUSING ASSEMBLIES. — An aluminum-alloy housing bolted to the nose section of the engine incorporates a mounting for the slip ring brush assembly, also called the brush cap. Quick removal of the brush holder from the housing is provided by trunk type latches. An electrical connector permits complete removal of the assembly from the airplane. The early type holder is mounted to the cap with vibration absorbing bushings. A later type holder of molded composition is attached directly to the cap without shock mounting.

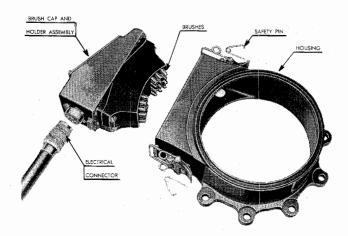
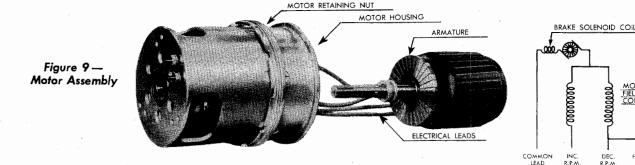


Figure 10—Brush and Housing Assemblies

i. UNIT CONSTRUCTION OF PROPELLER.—The unit type construction of the propeller greatly facilitates maintenance. The entire pitch change mechanism, known as the power unit, is readily interchangeable and can be removed and replaced as one compact unit. Spring-loaded electrical contacts in the speed reducer automatically complete the electrical connections between the motor and the hub when the power unit is attached to the hub. Blade assemblies can also be re-



moved and installed as an assembly with very little effort. This type of construction simplifies maintenance and also lends itself to convenience in handling and transporting propellers.

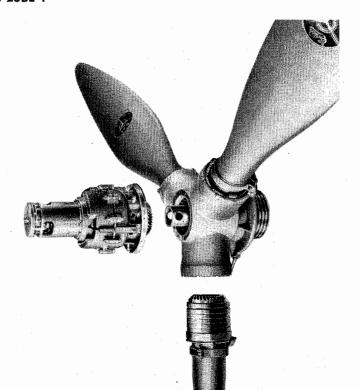


Figure 11— Unit Construction of Propeller

SECTION III INSTALLATION

1. GENERAL.

- a. Each propeller is fully assembled, balanced, and tested for proper functioning prior to shipment.
- b. The propeller is shipped with the propeller shaft nut, the front cone, the retaining snap ring, the grease seal and spreader, and the power unit removed. The power gear assembly is attached to the power unit.

2. INSTALLATION.

a. BLADES.

NOTE

To conserve shipping space it is sometimes desirable to remove two blade assemblies from the hub. If this has been done, the propeller must be reassembled in the manner described below:

- (1) Remove the hub with its one blade from the packing case and place the unit on any clean surface, the slip rings down. Although no table or spindle is necessary, such equipment, if available, will facilitate the work. (See figure 12.)
- (2) Thoroughly clean the bearings and sockets and coat with hub lubricant, grease Specification No. AN-G-4, Grade AA.
- (3) Place the gear backlash shims in the hub sockets with the chamfer toward the center of the hub.
 - (4) Make sure that the threads of the hub and

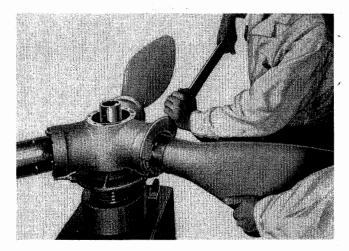
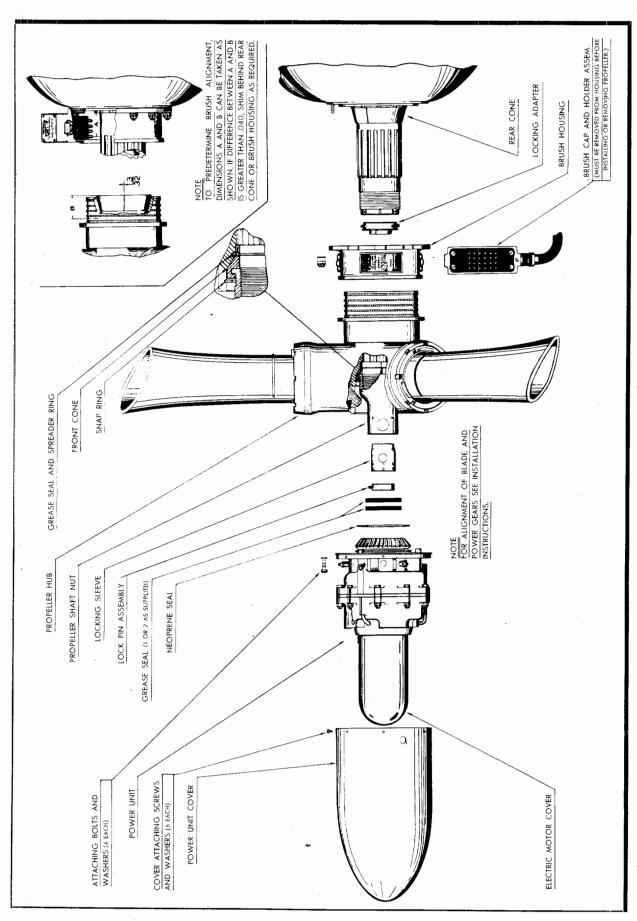


Figure 12—Assembly of Blade to Hub

blade nuts are free from metal chips and other extraneous matter and that the threads are thoroughly coated with a mixture of 70 percent white lead and 30 percent lubricating oil by volume (Specification No. AN-C-53).

- (5) Remove the locking plates which have been temporarily installed on the nuts of the loose blades.
- (6) Insert each blade into its proper socket. The blade assembly number is found stamped on the nut.





RESTRICTED

The hub socket number is stamped on the end of each blade barrel.

- (7) Screw each blade nut into the hub until one-half of the threads are engaged. Slide each blade assembly in and out sharply several times and finally pull it tightly against the blade nut. While exerting an outward pull on the blade, tighten the blade nut with the spanner wrench provided and a 10-pound brass hammer (or equivalent) until the paint marks on the nut and hub slots line up. If there are no paint marks, tighten the nut until it is barely possible to rotate the blade by hand.
- (8) Install the lock and safety wire the attaching screw.
- (9) Each propeller shipped disassembled is accurately balanced prior to shipment. A tag certifying this fact is attached to every unit and balancing is unnecessary if the blade assemblies are properly installed in their respective sockets. If the tag is missing or if there is any doubt as to the condition of the propeller, it should be balanced prior to installation merely as a precautionary measure.

b. PROPELLER. (See figure 13.)

- (1) Check thrust nut on engine for tightness; clean shaft threads and splines thoroughly, removing all nicks, burrs, and galls from the shaft and the face of thrust nut. Care should be taken to note that the threads on the shaft are not burred or pulled.
- (2) Remove nuts (and spacers or spacer ring if provided) from brush housing mounting studs on nose of engine and place brush housing on studs. Replace, tighten, and secure nuts.

NOTE

Except for the purpose of measuring for brush location as described below, the brush assembly must be left out of the housing until the propeller has been installed.

(3) In order to predetermine the location of the brush contact on the slip rings, it is necessary to measure as shown in figure 13. Measure the distance "A" between the face of the thrust nut and the center of the forward brush, and the distance "B" between the center of the front slip ring and the base of the rear cone which should be firmly held in the hub. If the difference between "A" and "B" is greater than .040 inch, shim behind rear cone or brush housing as required.

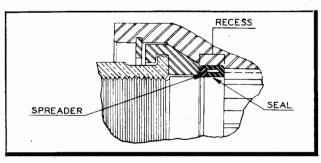


Figure 14—Section of Hub Showing Spline Recess for Grease Seal

- (4) Install rear cone on the shaft.
- (5) Place propeller shaft locking adapter in end of shaft.
- (6) Clean thoroughly and coat the threads of the propeller shaft and nut with an antiseize compound Specification No. AN-C-53. Apply a light coating of engine lubricating oil to the splines. The rear cone should be left dry.
- (7) To avoid the possibility of damaging the propeller shaft threads while installing the propellers, it is desirable to use a thread protector. To do this, proceed as follows:
- (a) Screw thread protector on shaft. Tighten by hand only.
- (b) Rotate blade assemblies (decrease pitch) until the cutaway portion of the blade gear is forward.
- (c) Being careful not to damage the shaft or rear cone seat, locate and slide the propeller halfway back on the shaft.
 - (d) Remove thread protector.
 - (e) Install grease seal and spreader.

NOTE

On earlier models, the splines are not recessed to take the grease seal and spreader. These types make use of a felt seal in the slot of the rear cone.

- (f) Install the nut, cone, and retaining snap ring. New cones may come in one piece, in which case it will be necessary to saw the halves apart and carefully remove the metal left in the split.
- (g) Slide the propeller back on the shaft and carefully start the nut on the threads of the propeller shaft. Tighten nut with a force of 250 to 300 pounds on a 3½-foot bar placed through the holes in the nut. Hammering on the bar is unnecessary and should be avoided.

NOTE

If a thread protector is not available, extreme caution should be taken not to damage the cone seats and shaft threads during installation.

(8) To make a final check of brush alignment on the slip rings, apply a light coating of Prussian blue on the ends of the slip ring brushes and place them in the housing. Rotate the propeller back and forth slightly and remove the brush assembly. The brush track, as indicated by the Prussian blue, should be in the approximate center of the slip rings and not closer than .020 inch to the slip ring insulators. Shim behind rear cone or brush housing as required to obtain proper alignment. When the alignment is satisfactory, clean the brushes and install the brush assembly, locking the latches with safety pins.

NOTE

Do not use cleaning solvent on the brushes.

(9) Fit the locking sleeve to the adapter inside the

propeller shaft nut so that a lockpin hole lines up with a hole in the nut. If the holes do not line up on the first application of the locking sleeve, remove locking sleeve and change its position on adapter until a hole does line up. Place lockpin assembly in position and release lockpin. Note that it passes through both the locking sleeve and the shaft nut.

c. POWER UNIT.

(1) Remove the power gear assembly from the power unit by removing the three countersunk attaching screws and the snap ring at the end of the power unit splined shaft. Check to see that the low limit switch arm is just riding on the low limit lobe of the cam. If it is necessary to operate the power unit to properly locate the cam:

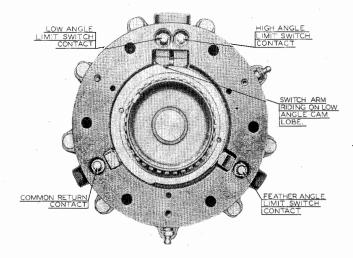


Figure 15—Limit Switches—R.H. Propeller

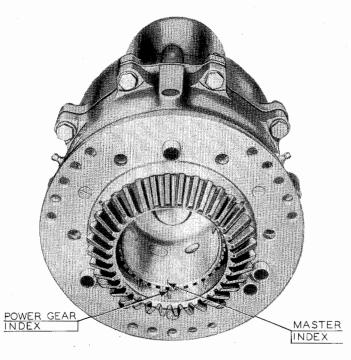


Figure 16—Indexing of Power Gear

- (a) Remove the steel mechanical stop plug, which is held in place by two 3/16-inch hex-head bolts just forward of the adapter plate. This must be done to eliminate the possibility of damaging the speed reducer by running into the mechanical stop when it is operated independently of the hub.
- (b) The cam may be rotated to the correct position by introducing a current through the proper two contacts on the face of the power unit.

NOTE

The power unit splined shaft has a master spline which is indicated by a mark on its end. The power gear is marked in increments of one degree. A decalcomania or etched lines on the blade when aligned with an index on the front of each blade socket indicate the blade angle. The blade angles of propellers are covered in Section VII.

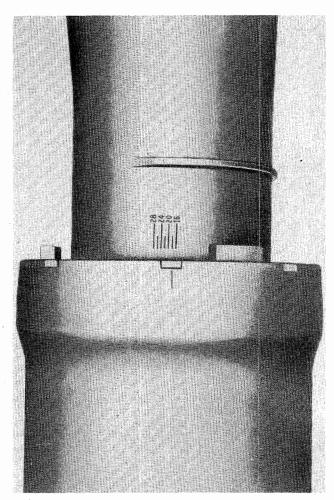


Figure 17—Indexing of Blade

- (2) Replace the power gear, having the master spline in line with the mark indicating the desired low blade angle, then replace the three attaching screws and the snap ring.
 - (3) Rotate the blade assembly until the index on

the blade indicates the desired low blade angle (same angle as the power gear).

- (4) Install felt grease seal or seals as supplied over propeller shaft nut and make sure that the nut is satisfactorily tightened and locked.
- (5) Clean contacts on the face of hub and power unit.

NOTE

Do not use cleaning solvent on the contacts. Dirt may become lodged between the rattle contacts and must be blown out.

- (6) Place the grease seal on the adapter plate. Place power unit on hub, aligning the contact points and bolt holes of the power unit hub. Push unit hard against the hub so that the power gear meshes with the blade gears. Secure the unit tightly to the hub with six attaching bolts and safety wire them.
- (7) Lubricate as outlined in section V, paragraph 3.b.
- (8) Replace mechanical stop plug in the proper position as indicated by the mark "O" stamped on the stop and housing. Inspect the blade shank indices to see that the blades are at the correct angle.

SECTION IV

1. PRINCIPLES OF OPERATION.

a. The blade angles of the Electric Propeller are controlled by means of a reversible electric motor. The electrical energy required for operating the motor is taken from the airplane power supply and passes through brushes mounted in a housing attached to the engine nose to slip rings which are fixed to the rear of the propeller hub. From the slip rings, the electrical energy passes through connector leads in the hub to leads in the speed reducer and thence to the motor. The motor changes the blade angles through a two-stage, planetary-type speed reducer which drives a master bevel gear. This gear meshes with a bevel gear attached to the shank of each blade. Thus, depending upon the direction of rotation of the motor, the angle of the blades is increased or decreased.

- b. A brake is attached to the front end of the electric motor for the purpose of stopping its rotation when the pitch changing current is cut off. The brake also provides a definite lock which holds the blades in a fixed position when no angle change is desired.
- c. Switches are provided in the hub end of the power unit to limit the low and high blade angles of the flight range and also to halt the pitch change at the feather setting. These switches are cam operated and are connected into their respective motor leads. Reverse thrust propellers have an additional switch which limits the angle to that required for such operation.

2. OPERATING INSTRUCTIONS.

For operating instructions, refer to the Handbook of Flight and Operating Instructions applicable to the airplane.

SECTION V

SERVICE INSPECTION, MAINTENANCE, AND LUBRICATION

1. SERVICE TOOLS.

For the required service tools, refer to T. O. No. 03-20B-15.

2. INSPECTION AND MAINTENANCE.

Column No. 28-Propellers and Accessories.

NOTE

In accordance with Technical Order No. 00-20A-2, a summary of periodic inspections prescribed herein will be entered in the Master Airplane Maintenance Instruction Form maintained in the back of form 41-B for the airplanes affected.

Preflight.

The mechanic will perform the following check: The circuit breaker must be in the "ON" position. Engines may be started and warmed up with selector switch in "AUTOMATIC." To check selective fixed pitch control operation, open throttle to turn engine 1,000 to 1,200 rpm. Hold selector switch in "DECREASE RPM" position until a reduction of not more than 200 rpm in engine speed is noted, then hold in "INCREASE RPM" position until original engine rpm is obtained. When the engine rpm ceases to increase, the propeller has reached its low blade angle.

To test constant speed operation, place selector switch on "AUTOMATIC" and place the cockpit control lever in "TAKE-OFF" position. Open throttle until engine turns approximately 70 percent rated rpm, and pull the cockpit control lever back until a reduction of approximately 200 rpm in engine speed is noted. Return cockpit control lever to "TAKE-OFF" position,

noting that original rpm is resumed. Individual magnetic operation can be checked while the selector switch is in the "AUTOMATIC" position.

Pilots are urged to make the constant speed check given above prior to all take-offs and to regard this check as supplementary to the routine checks made by the ground crew.

NOTE

This preflight operation check will serve as added insurance against making a take-off with an inoperative propeller, which may be caused by an incorrect switch setting or possible mechanical difficulty.

Daily.

NOTE

The following instructions cover servicing of only the propeller. For control system maintenance, refer to Handbook covering the type of control system used.

Thoroughly clean and visually inspect hub and blades for damage or defects that may have occurred during previous operation. The blades are to be cleaned with a light lubricating oil and their inspection should include a careful check of the surfaces for damage such as bends, cracks, dents, and nicks. The full length of the leading and trailing edges as well as any dents or scars on the blade surface should be carefully examined for cracks. Each questionable scratch should be carefully examined to determine whether it is a scratch or crack. Raised edges of scratches, nicks, etc., which are likely to induce cracking, should be dressed off with a hand stone or emery cloth.

Test operation of the pitch changing mechanism by placing the selector switch in "DECREASE RPM," then "INCREASE RPM" position to check operation in both directions. On multi-engine planes, place feather switch to "FEATHER" position to determine that the propeller will feather and then return the switch to "NORMAL" position. On those installations employing reverse thrust operation, this circuit may be tested by placing the reverse safety switch to "ON" and the reverse switch to "REVERSE." The warning Tel-lite should be on until the reverse blade setting is reached. Return to normal by moving the reverse switch to "RETURN." The Tel-lite should be on until the low pitch setting is reached. As soon as the Tel-lite is off, the reverse switch must be moved to "NORMAL" and the reverse safety switch to "OFF."

25-Hour.

Remove the brush cap assembly from the housing by releasing the two latches. Inspect brushes and slip rings for wear. Brushes will be cleaned by wiping with a dry cloth. No solvent will be used on the brushes since they are impregnated with oil which is necessary for proper wearing characteristics. Cleaning solvent will dissolve this oil and accelerate wear of the brushes and slip rings. Wipe slip rings by holding cloth against the ring while propeller is being rotated. Brushes of the molded type holder are inscribed on the side with an arrow which points toward the front of the brush. When the point of wear reaches the tip of the arrow and beyond, the brush must be replaced. Brushes of the early type holder should extend 3/8 of an inch from the block. If they do not extend this distance, examine for worn brushes, permanently compressed springs, and dirty brush guides.

On multi-engine planes, place feather switch to "FEATHER" position to determine that the propeller will feather and then return the switch to "NORMAL" position.

100-Hour.

Remove the brush assembly from the housing by releasing the two latches. Remove block assembly from the cap, remove cover assembly, and remove brush assemblies from the block. Clean oil and carbon dust from the brush holder. Clean out brush guides with a small spiral hair brush or by drawing a cloth through. Brushes themselves will be cleaned by wiping with a dry cloth.

NOTE

Do not use cleaning solvent on the brushes.

Brush blocks of the molded type may be made arcresistant to prevent shorting between metal brush guides by the application of varnish to the concave surface of the brush block. To accomplish this, sand the surface with number OO sandpaper until a smooth surface is obtained and scrape the ends of the brush guides clean. Preheat the block at 163°C (325°F) for a minimum of one hour. While the block still retains a heat of approximately 65.5° to 93.3°C (150° to 200°F) hold the concave surface in a downward position and brush on one coat of varnish. Inspect inside surfaces of brush guides for presence of varnish. Air dry the block for a minimum of one hour and place (concave surface downward) in an oven. Bake for two hours at 163°C (325°F).

Remove motor cover. Inspect the general condition of the electric motor. Check the brake clearance by using a feeler gage between the brake plate and the inner ring of the solenoid housing. The clearance should be between .010 to .020 inch. Clearance adjustments may be obtained by adding or removing shim laminations on the armature shaft, directly behind the brake disk assembly. After inspection, replace motor cover and safety.

3. LUBRICATION.

- a. Lubricate the hub with grease, Specification No. AN-G-4, grade AA, soft, using a pressure gun on the Zerk fitting located on the speed reducer housing just forward of the front hub face until hub is full, as indicated by a flow of grease from the relief fitting.
- b. Check the oil level in the speed reducer by removing the filler plug located near the front of the housing and rotating the propeller until the plug opening is

approximately 20 degrees below the horizontal plane when the airplane is at a ground angle of approximately 12 degrees, or 8 degrees below the horizontal when the

airplane is leveled. The oil in the speed reducer should then be at the plug opening. If not, fill the gear assembly at this point, using oil, AAF Specification No. 3600.

SECTION VI DISASSEMBLY, INSPECTION, REPAIR,

AND REASSEMBLY

1. OVERHAUL TOOLS.

For the required overhaul tools, refer to T. O. No. 03-20B-15.

2. DISASSEMBLY.

a. GENERAL.—No propeller overhaul interval other than that specified by the operator is recommended, since the type of operation to which the propeller is subjected is the principal determining factor.

b. POWER GEAR.

NOTE

Before removing the power gear assembly from the power unit, mark or note the spline space on the power gear mating with the mark on the power unit splined drive. This indicates the low angle setting of the propeller.

- (1) Remove the power gear assembly from the power unit by removing the three countersunk attaching screws and the snap ring from the splined end of the low speed bell gear.
- (2) Remove the gasket and the insulator-lined steel bushings from the rear of the speed reducer housing.
- (3) Press power gear, shims, and grease seal from the adapter plate, being careful not to damage the grease seal or the shims. Remove the snap ring and bearing.

c. BRAKE AND MOTOR ASSEMBLY.

- (1) Remove motor cover.
- (2) Unscrew the nut on the front end of the motor shaft and, by using a brake puller, remove the brake disc assembly. Then remove the brake disc locating key from the motor shaft.
- (3) Disconnect the solenoid terminals and remove the three attaching bolts that secure the brake diaphragm assembly and solenoid.
- (4) Unscrew the motor retaining ring nut from the front speed reducer housing.
- (5) Pick the insulation from around the four motor terminal screws at the rear of the speed reducer housing and remove the screws, thereby disconnecting the motor leads.
- (6) Using a puller, remove the motor from the speed reducer housing; at the same time, work the motor leads out of their passages in the housing.

NOTE

Extreme caution should be taken to prevent any side loads on the armature which might

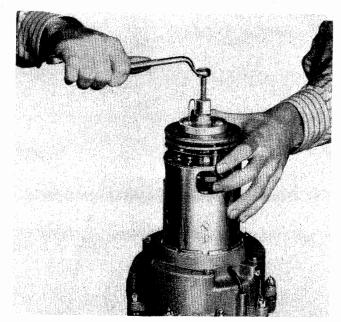


Figure 18—Removal of Brake Disk

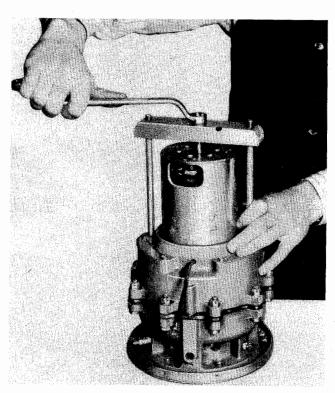


Figure 19—Removal of Motor

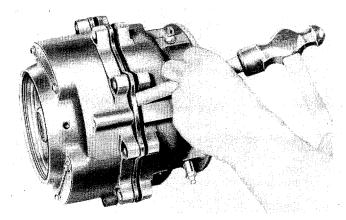


Figure 20—Separation of Housing

bend the shaft or crack the sleeve extending through the speed reducer oil seal.

- (7) Press the armature from the motor housing.
- (8) Press bearing from the motor housing.
- (9) Remove the brushes from the brush holder.

d. SPEED REDUCER.

- (1) Remove oil filler plug from speed reducer housing and drain the oil.
- (2) Remove the nuts and bolts which hold the front and rear housings together.
- (3) Using a hammer and a brass or wooden drift, tap the fixed ring gear from rear housing. The ring

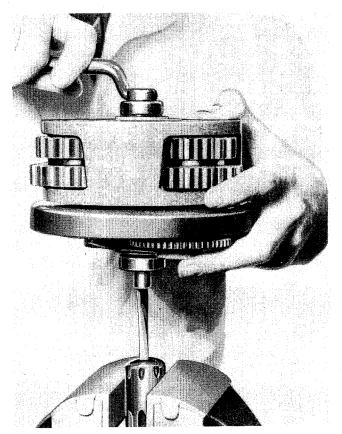


Figure 21—Removal of Drive Gear Nut

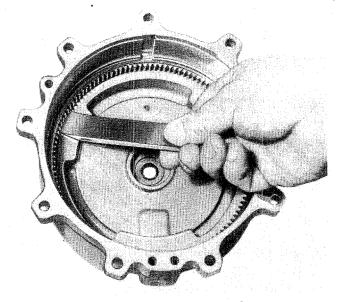


Figure 22—Removal of High Speed Fixed Ring Gear

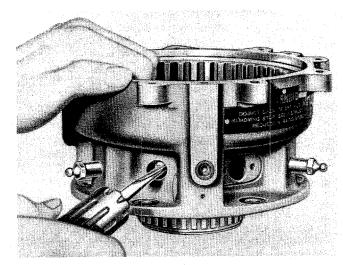


Figure 23—Removal of Mechanical Stop Segment

gear, the front housing, and the gear assembly will then separate from the rear housing.

- (4) Remove the gear assembly from the front housing. Disassemble it by removing the nut from the end of the shaft and removing gears, spiders, bearings, and spacers. Remove the gears from the high and low speed spiders.
- (5) The front oil seal need only be removed if it is defective as ascertained by visual inspection.
- (6) Remove the fixed gear snap ring from the front housing and carefully work out the fixed ring gear.
- (7) Rotate low speed bell gear in the rear housing until cut-out cam and mechanical stop screws are visible through the breather holes. Insert a proper size screw driver and remove the screws. The screws holding the segment turn to the right when being removed and fall into the bell gear splined hub.

- (8) Remove the cam attaching screws and then remove the cam by making use of two 10-32 screws, two inches long, as jack screws in the tapped holes provided in the cam.
- (9) Push the bell gear out of the rear housing and remove the bearing, oil seal, and also the felt grease seal.

e. LIMIT SWITCHES.

- (1) Remove the terminal studs from the limit switch assemblies.
 - (2) Remove clevis pins and switch arms.
- (3) Remove limit switch assemblies from rear housing. The switch assemblies need not be disassembled further.

f. REMOVAL OF BLADES FROM HUB.

- (1) Place the hub assembly slip rings down on a spindle.
- (2) Remove the locks from the blade nuts, marking location of same on the hub and nut with paint. Unscrew the blade retaining nuts from their sockets.
- (3) Remove each blade from its socket and place it on a clean assembly table with the bearings over-hanging the edge.

g. CUFFS.

- (1) Carefully tap the hinge pin out of the hinge and remove all cover attaching screws.
- (2) Remove the trailing edge stiffener by driving out the dowel pin.
- (3) Remove the two nuts which hold the support in place.

b. BLADES.

NOTE

Less trouble will be experienced when the propeller is reassembled if the individual parts of each blade assembly are kept together.

- (1) Slide the retaining nut back over the blade shank.
- (2) Using brass or aluminum blocks to cushion the blows, tap the bearings backward until the blade gear locking pin is accessible.
- (3) By using a pin puller, remove the pin from the blade.
- (4) Unscrew the gear from the blade and remove the bearings and nut. If the gear does not turn freely place it in a copper-jawed vise and rotate the blade with a paddle.

i. HUB.

NOTE

It is necessary to disassemble the hub only when the slip rings need replacing. It is not necessary to remove slip rings for magnaflux inspection of the hub.

- (1) Remove the screws holding the slip ring assembly to the hub.
 - (2) Remove the slip ring assembly with its con-

- nector rods from the hub, taking care that rods are not damaged.
- (3) Unscrew the connector rods from the slip ring assembly and remove their insulating tubes.
- (4) Mark the slip rings so they can be reassembled in the proper order and position, then slide them off the insulating sleeve.

. j. BRUSH HOLDER.

- (1) Remove the warning plate.
- (2) Disconnect the wires from the brush terminals.
- (3) Unscrew the four screws which secure the brush holder to the mounting bushings in the brush cap and remove the holder.
- (4) Remove the two screws which hold each brush terminal block in place. On the molded type holder it is only necessary to remove the cover.
 - (5) Remove the brushes from the holder.
- (6) Unscrew the four attaching screws and remove the electrical connector socket.

3. INSPECTION AND REPAIR.

a. GENERAL.

(1) After disassembly all of the propeller parts should be inspected according to the following outline and necessary repairs and replacements made. Before inspection, the parts should be cleaned thoroughly and oiled to prevent oxidation.

CAUTION

DO NOT USE GASOLINE CONTAINING TETRAETHYL LEAD IN CLEANING.

- (2) The hub, retaining nuts, gears, gear pins, and all other moving steel parts should be inspected by means of the magnaflux method.
- (3) All bearings should be inspected for possible damage, wear, or roughness.

CAUTION

If one active bearing of a stack of blade bearings is found to be defective, all active bearings of the stack must be replaced. If the preload bearing is defective, it may be replaced provided that its inner race is not integral with that of the adjacent active bearing. If the inner race of the preload bearing is integral with that of the adjacent active bearing, the entire bearing stack must be replaced.

- (4) All gears must be inspected for excessively worn or damaged teeth.
- (5) All threads shall be free of nicks, distortions, wire edges, and roughness.
- (6) All parts on which the plating is worn or corroded must be replated. During the replating operation, care must be taken not to plate bearing surfaces or threads.

b. MAGNETIC BRAKE.

(1) Test the solenoid with an ohmmeter or megger

for an open circuit or weak insulation. Test solenoid for resistance using the following table of specified resistance limits. Check for damaged coil terminals. If a brake winding is defective, it is desirable to replace the entire housing assembly, as installation of the winding in the housing is a highly specialized operation.

Assembly No.	Voltage	Specification
101379-1	12V	.029 to .035 ohms
101379-2	24V	.117 to .143 ohms
108665-1	12 V	.035 to .043 ohms
108665-2	24V	.117 to .154 ohms
108065-1, -3	12V	.036 to .042 ohms
108065-2	24V	.126 to .154 ohms

(2) A glazed surface on the brake facing may be removed with sandpaper. Badly worn facing should be replaced. Inspect springs to make sure that they are not permanently compressed. Inspect diaphragm for possible cracks or loose rivets.

c. ELECTRIC MOTOR.

- (1) Check trueness of armature shaft in a lathe.
- (2) If the commutator of the armature is cut or burned it should be resurfaced in a lathe and the mica undercut to a depth of approximately one-sixteenth inch.
- (3) Try the armature circuits with a growler or light and visually inspect the windings for broken insulation. If the insulating varnish on the armature is chipped or deteriorated, clean thoroughly and apply two coats. Inspect the serving of the winding leads to the commutator of the armature. If there is any sign of looseness, the serving must be replaced and thoroughly coated with insulating varnish.
- (4) Any damage or excessive wear of the spline or keyway and tang of the armature shaft is cause for replacement of the armature.
- (5) If a motor brush is less than one-half inch in length, replace the brush and spring assembly.
- (6) Check tightness of brush rigging. To eliminate any movement, straighten three retaining bolt lock washers and tighten brush holder retaining bolts. On early type motors the brush rigging assembly is attached by means of flathead screws mounted through the motor case. These screws should be inspected for tightness and the general condition of the brush rigging noted. Inspect motor leads for broken or damaged insulation and splice on new leads if necessary. Damaged insulation may be served with linen cord and coated with insulating varnish.
- (7) Carefully inspect the motor housing and front speed reducer housing for signs of chafing. Inspect locking key on motor cover for loose rivets.

d. SPEED REDUCER.

- (1) Inspect all bearing seats to be sure that the outer bearing races do not rotate in the bearing seats.
- (2) Inspect the housings and spiders for possible cracks or damage.

- (3) Inspect the cam lobes and limit switch arms for wear or damage.
- (4) If switch contacts are corroded or pitted they should be smoothed with a stone. If it is found necessary to replace the contact or the contact spring, the whole contact assembly should be replaced.
 - (5) Replace worn grease and oil seals and gaskets.
- (6) Inspect wire connections for broken or damaged insulation.
- e. POWER GEAR.—Should the grease seal be deteriorated or damaged, it should be replaced.

f. HUB

- (1) A slight amount of galling, and scratches caused by removal of blade assemblies may be found in the blade sockets. Unless excessively deep, these need not be cause for rejection. Removal of metal should be held to a minimum. Remove only raised edges and radius the bottom of scratches with a stone.
- (2) Inspect the cone seats. If metal has been picked up from the rear cone, it should be cleaned out and the seat lapped with a fixture made for that purpose.
- (3) If it is found necessary to replate the hub, protect the cone seats, threads, and ground surfaces of the blade sockets against plating.

g. SLIP RINGS.

- (1) Inspect all slip rings for excessive wear. They should be replaced if the outside diameter is less than 6.094 inches (No. 50 SAE spline shaft) after they have been turned and smoothed as outlined in section VI, paragraph 3.a.(7).
- (2) Inspect connector rods for damaged threads and contact points for smoothness. The contact points can be smoothed with a stone.

b. BRUSH HOLDER.

- (1) The brushes should have no frayed or broken leads and should slide freely in the holder.
- (2) The brushes of the molded type holder must be replaced if the point of wear has reached or gone beyond the tip of the arrow inscribed on the brush. The brushes of the laminated type holder should extend three-eighths inch from the block. If they do not extend this distance, examine for worn brushes or permanently compressed springs.
- i. BLADES.—With the exception of the removal of small nicks and bruises, as outlined below, all repair of blades should be accomplished by an authorized depot or agency.

(1) CARE OF BLADES AT OVERHAUL.

(a) All sharp, raised edges of scratches and nicks must be dressed off with a hand stone or emery cloth. However, small shallow dents which are smooth and have no sharp corners need not be removed. In reworking nicks, all edges should have small radii to conserve metal in the damaged section. This procedure varies from that employed on dural blades where the nick may be blended into the contour of the blade without weakening it structurally.

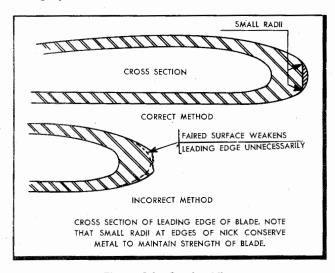


Figure 24—Section View

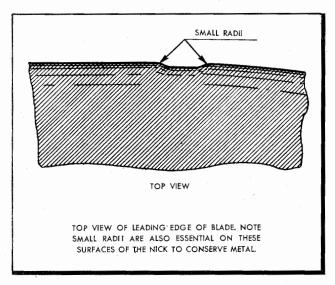


Figure 25—Top View

- (b) If it is necessary to remove more than the following amounts of metal to repair defects, the blade is unsafe and must not be used for flight.
- 1. As seen in figure 26, the blade may be divided into three regions: region "A," the solid metal of the leading edge; region "B," the solid metal of the trailing edge; and region "C" which includes the hollow portion of the blade and extends one-eighth inch into the solid portion of the blade.
- 2. For the outer two-thirds radius of the blade, metal can be removed in region "A" to a depth of .125 inch and in region "B" to a depth of .190 inch without impairing the strength of the blade. For the inner one-third radius of the blade, metal can be removed in regions "A" and "B" to a depth of .030 inch. The number and relative location of repairs made to a maximum allowable depth should be considered in deciding the serviceability of a blade.
- 3. When a nick or bruise is to be removed in region "C," great care must be exercised so that metal

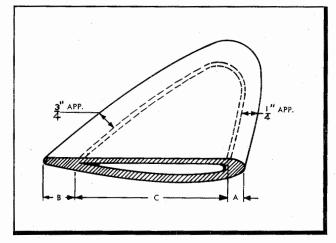


Figure 26—Cross Section of Blade

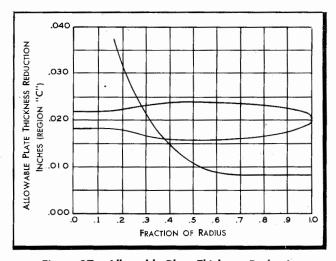


Figure 27—Allowable Plate Thickness Reduction

is not removed to a depth greater than is indicated on the chart. A record should be kept of any considerable removal of metal from this region for the purpose of future reference.

- 4. If the blade is corroded or shows signs of oxidation, the affected areas may be polished with steel wool or fine emery cloth. Periodic coating of the blade with Specification No. AN-C-52 (Whiz Compound) will suffice to prevent further corrosion and oxidation.
- (c) At major overhaul, or whenever their condition is doubtful, blades shall be inspected by means of the magnaflux system. This inspection should be accomplished as follows:
 - 1. Remove paint from blade.
- 2. Mount the blade in a magnaflux machine. Should the equipment be too small to accommodate the blade, clamps connected to the power supply may be attached to the tip and butt of the blade. Extreme caution should be taken to insure proper electrical contact so that the blade will not be burned. In order to insure against a large current drop, the leads to the contact clamps should be as short as possible and the diameter of the connector should be approximately five-eighths inch.



Figure 28—Magnaflux Inspection

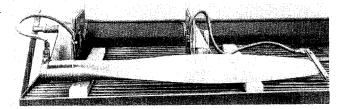


Figure 29 - Magnaflux Inspection

3. Using a power supply of 3,000 to 4,000 amperes at 6 volts, magnetize the blade by rapidly making and breaking the circuit 2 or 3 times.

NOTE

Best results can be obtained if the magnaflux solution is poured over the blade at the same time that the blade is being energized.

- 4. Examine the entire blade, especially the leading and trailing edges in the area of the weld, the pinhole, and the threads. Any crack is cause for immediate rejection of the blade from service.
- 5. Upon completion of the magnetic inspection, demagnetize the blade.
- (2) REJECTION OF BLADES FROM SERVICE.

 -A blade which has any of the following defects shall be immediately removed from operation:
 - (a) A crack of any kind.
 - (b) A bend of any description.
 - (c) A bullet hole.
- (d) A deep cut or scar that is likely to induce cracking.
- (e) A nick that should be removed to prevent the blade from cracking, but the removal of which would result in the blade's failing to meet the minimum thickness requirements as listed in paragraph 3.i.(1)(b) of this section.
- (f) An out of track condition that exceeds permissible limits.

With the exception of (a) and (c), it is possible to repair the defects listed above if they are not too severe.

j. CUFFS.

- (1) Inspect the support, stiffener, and cover for possible cracks.
- (2) Make sure that all rivets are tight and that the chafing strips have not deteriorated.

k. BLADE RETAINING PARTS.

- (1) Inspect bearings as outlined in paragraph 3.a.(3) of this section. Wash bearings carefully and pack with hub lubricant Specification No. AN-G-4, Grade AA.
 - (2) Inspect the blade nut threads for smoothness.
- (3) Note that the grease seal has not been damaged or deteriorated and that the garter spring is in place.

1. PROPELLER SHAFT ATTACHING PARTS.

- (1) Inspect the propeller shaft nut seal or seals and replace if deteriorated.
- (2) Inspect propeller shaft seal and spreader for possible damage and replace if necessary.
- (3) Inspect the propeller shaft nut locking adapter and locking pin assembly for wear or damage. Make sure that the plungers are operating freely.

4. REASSEMBLY.

a. HUB.

- (1) Assemble the slip rings and insulator rings on the insulator tube.
- (2) Install the labyrinth seal on the front end of the slip ring mounting sleeve and coat the joint with Copaltite or some similar suitable sealing compound.
- (3) Install the six insulator bushings into the holes provided for the attaching screws.
- (4) Insert the long insulator bushings in place. Coat the threads of the connector rods with litharge and screw each into its proper slip ring.
- (5) After coating both sides of the slinger ring or spacer with a sealing compound, mount it on the slip ring assembly and press the assembly in place on the hub. Tighten and lock wire the six attaching screws.
- (6) Check the contact points to be sure they are either flush with, or not exceeding minus .005 inch from, the front face hub. To adjust, add or remove shims directly under head of removable contact. On final setting, coat threads of contact with litharge.
- (7) Mount the hub assembly on a balancing mandrel or an equivalent form of arbor that will run true in a lathe. With a dial indicator, check the trueness of the mandrel in the lathe centers. Cone seats should be clean and smooth before mounting the hub on the mandrel. Rotate the hub by an angle bolted to the face plate of the lathe, so as to engage with several power unit attaching bolts screwed into the hub.
- (8) With a straight cutting tool approximately one-sixteenth inch wide, remove sufficient material from the surface of each slip ring to remove all irregular surfaces. The final cut should be made very lightly and with a very slow feed. Care should be taken not to remove more material than is necessary since the slip ring must be replaced if its outside diameter is less than 6.094 inches (SAE No. 50 spline shaft propeller). The diameters of the slip rings need not be alike but must be concentric with the cone seats within .003 inch total indicator reading. A strip of coarse cloth

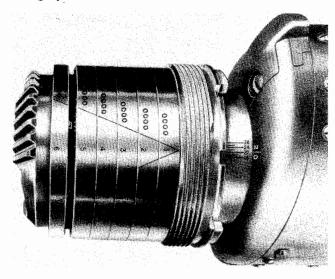


Figure 30—Blade Bearing Assembly

such as the back of emery cloth may be used for polishing the rings. Check and, if necessary, true up slip ring separators in a like manner, removing as little material as possible to bring concentricity within .005 inch total indicator reading. The distance from the slip rings to the top of the separators shall not be less than .060 inch. The same cutting tool can be used for both slip rings and slip ring separators.

b. BLADES.

- (1) Place the blade on a clean assembly bench with the shank overhanging the edge.
- (2) Place blade nut (with its grease seal and garter spring) and blade bearings on the shank of blade in order named.

NOTE

It is essential that the bearings of a stack be kept together. Individual bearings of stacks, copper plated on the outer diameter, may be identified by means of a serial number etched on the shoulder of the outer race of each bearing. This serial number is the same for all bearings in a stack. The serial and individual bearing number of those stacks which are not copper plated are etched on the outer diameter of each bearing.

(3) Screw the proper blade gear into each blade. Each gear is stamped with a number which should correspond with the number stamped on the shank end of its blade.

NOTE

In the interest of efficiency, serviceability, and maintenance, it has been found necessary to change several details of the blade assemblies. The most important change made was in the design of the pilot used on the blade gears.

(a) The pilot on the early type gear was tapered. This gear was individually fitted to its blade before the

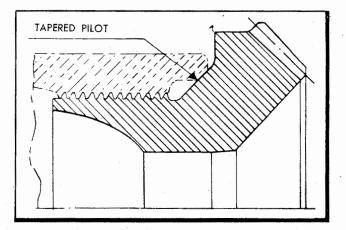


Figure 31—Tapered Pilot Gear

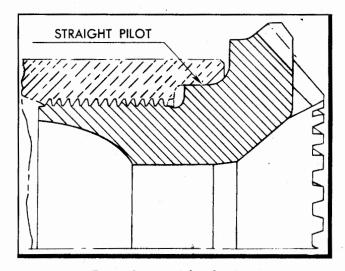


Figure 32—Straight Pilot Gear

pinhole was drilled and reamed and is, therefore, not interchangeable. Although replacement is usually a factory operation, this gear may be fitted and drilled by a shop which has the proper equipment and approved fixtures.

- (b) The later type gear, however, has a straight pilot and is completely interchangeable with other gears bearing the same part number. The pinhole of a replacement gear must be checked while it is in the blade by using a fixture approved by the manufacturer. Should this check prove that the pinholes of the gear and blade are not in perfect alignment, or not of the same size, the pinholes must be reamed to the next oversize as described in paragraph (4) below.
- (c) Upon assembly, taper pilot gears are preloaded by tightening the gear against the shoulder of the blade until the pinholes line up. Straight pilot gears are located axially by hand and preloaded with proper tool.
- (d) It is imperative that exact pinhole alignment be obtained in order that service failures may not occur. The pinholes must be free from all metallic pick-up. If a hole has picked up slightly, it may be

cleaned with crocus cloth or emery paper. If a pin has picked up, it should be replaced. The pin fit must be from .0000 inch to .0005 inch tight.

- (4) If it is found advisable to use an oversize pin, proceed as follows:
- (a) Place the checking fixture on the blade shank and tighten the assembly bolts so that the fixture can be moved freely but without play.
- (b) Locate the fixture bushing visually concentric over the blade shank hole.
- (c) Insert the tapered blade shank locating pin and press steadily downward until it is no longer possible to move, in either direction, the fixture on the blade.
- (d) Tighten the fixture bolts firmly but not so much as to cause the gear to bind.
- (e) Remove the tapered pin and locate the gear hole visually concentric with the blade shank hole.
- (f) Insert the blade gear locating pin. This pin will permit a slight amount of backlash in the gear which can be measured by a dial indicator attached to the fixture. Rotate the gear to the extreme left and move the dial indicator to zero. Then rotate the gear to extreme right and note indicator reading. By rotating the gear to the left one-half the total indicated reading, the blade shank and gear holes will be perfectly aligned.
- (g) Preload the gear with a torque force of 50 foot-pounds applied to the screw of the proper preload tool, part No. ST1074.
- (b) Insert the proper size reaming bushing in the fixture and ream blade and gear holes to one size above the existent blade shank hole size. Use following table in selecting correct oversize pin and reamer.

Size of Pin	Next Over-	Reamer
Removed	size Pin	Size
	10101/1	+.001
.3755	101814-1	3815000 + .001
.3825	101814-2	.3901000
,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		+.001
.3911	101814-3	.3980000

- (i) Remove the fixture and preload tool.
- (j) Remove the gear and remove all pick-ups and rough edges from around the blade shank and gear holes.
- (5) Install the blade nut and bearings as described in paragraph 4.b.(2).
- (6) Install the gear and enter the proper size pin (threaded hole outward) in the blade shank hole. Tap the pin very lightly until the gear cannot be rotated in either direction.
- (7) Preload the gear with a torque force of 65 to 70 foot-pounds and drive the pin flush or below the blade shank surface.
- (8) Slide the bearings and blade nut against the blade gear.
 - (9) Make certain that the gear plug and gear mat-

ing surfaces are absolutely clean and dry. Then, using a soft metal drift with a fiber adapter facing to protect the anodized plating of the gear plug, drive the plug securely into the gear.

c. BLADES IN HUB.

- (1) With the slip rings down, place the hub on a propeller table or some other suitable spindle.
- (2) At this time, carefully reinspect the hub barrel and blade nut threads for wide edges, pick-ups, and other imperfections, making sure that they are absolutely clean. Coat the blade sockets and bearings with Specification No. AN-G-4, Grade AA; coat the threads of the blade nuts and barrels with a white lead mixture Specification No. AN-C-53 (70 percent white lead and 30 percent lubricating or castor oil by volume).
- (3) Place a shim in the bottom of each blade socket. The chamfer side of the shim should face the center of the hub.
- (4) Insert each blade assembly in its proper socket. Usually the blade with the lowest manufacturing number goes into socket No. 1, the next highest into socket No. 2, the highest numbered blade into socket No. 3.
- (5) Screw each blade nut into the hub until one-half of the threads are engaged. Slide each blade assembly in and out sharply several times and finally pull it tightly against the blade nut. The nut must be tightened by striking the wrench with a 10-pound brass hammer until the nut and hub locking slots, which have been marked during disassembly, line up. When properly assembled, it will barely be possible to rotate the blades by hand. However, a check should be made with a wooden blade wrench to make sure the blades will rotate.
- d. CUFFS.—Each cuff assembly has a serial number which is stamped on the support, the stiffener, and the cover. Since these parts are not readily interchangeable every part of the cuff must bear the same serial number.
- (1) Place the support around the shank of the blade with the groove of the support over the blade shank shoulder.
- (2) On those supports secured by two bolts, insert the bolts so that their heads face the flat side of the blade. On the supports secured by a bolt and stud, insert the bolt so that its head faces the hub.
- (3) Line up the 20-degree mark stamped on the camber half of the support with the corresponding degree mark on the blade shank index.
- (4) Tighten the two bolts, or bolt and stud by applying a torque of 50 to 75 inch-pounds.
- (5) Place the stiffener in place. Tap the dowel pin into place until it is centered between the halves of the support.
- (6) Attach cover. It may be necessary to clamp or strap the cover in place in order to line up the screw holes.
- (7) Tap the hinge pin in place, making certain that the flat side of the pinhead is against the stiffener.

When checked with a feeler gage between the bearing surface of the chafing strip and the blade surface, the cuff must fit within the following dimensions:

e. SPEED REDUCER.

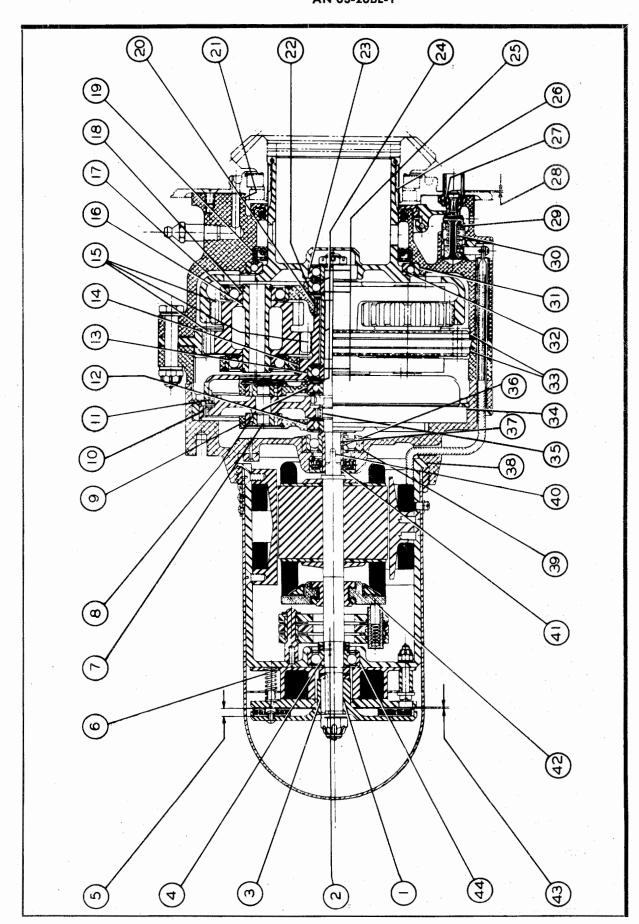
- (1) Install oil seal and felt grease seal in the rear housing.
- (2) Place the mechanical stop segment in the breather section of the rear housing.
- (3) Press the low speed bell gear bearing onto the shoulder of the bell gear, so that the thrust face of the outer race is away from the bell gear.
- (4) Press bell gear into place in the rear housing, being careful to avoid damage to the oil seal. A slightly tapered sheet-metal sleeve, fitting over the splines, may be used to guide the spline shaft through the oil seal.
- (5) From the inside of the bell gear, insert the two screws which hold the stop segment in place. Tighten the two screws by inserting a screw driver through a breather hole in the housing. The screws should be staked in place.
- (6) Install cut-out cam by pressing it into place and installing the screws and lock washers through the breather holes in the housing. The smaller attaching lug of the cam fits into the recess in the mechanical stop segment.
- (7) If removed, press the oil seal in front housing and stake in place.

- (8) Install high speed fixed ring gear with its keys and retaining snap ring in front housing, making certain that the snap ring is in such a position that keys can not work out.
- (9) Assemble both the high and low speed spider assemblies.
- (10) Assemble the internal gearing having the index marks on the planet gears lined up on the center lines of the spiders. Make sure that the nut on the end of the shaft is tight and cottered so as not to interfere with the bearing or housing.
- (11) Using a new gasket (shellac unnecessary), install the low speed fixed ring gear in place on the rear speed reducer housing. While turning the gear assembly, mesh it in place in the rear housing. If the gears have been properly indexed, they will mesh with the low speed bell gear without difficulty.
- (12) Insert a second gasket (shellac unnecessary) on the top of the fixed ring gear and install the front housing, being careful to guide the shaft through the front oil seal without damaging it.
- (13) Place the bolts which hold the front and rear housings together into position so that the head of each bolt faces the rear of the speed reducer. Install and tighten the nuts, applying a torque of 180 to 250 inch-pounds. While tightening, turn speed reducer with screw driver. If the unit binds, it should be disassembled and checked for improper assembly.
- (14) Pour one pint of speed reducer oil AAF Specification No. 3600 through the plug opening in the front housing and secure plug.

LIST OF TOLERANCES-POWER UNIT ASSEMBLY

Reference			
Number	Description	Max.	Min.
1	Fit of brake disc on motor shaft—diameter	.005	.000
2	Motor shaft end play	.010	.005
3	Fit of key in motor shaft and brake disc-side clearance	.0020L	.0005T
4	Fit of bearing on motor shaft-diameter	.0005L	.0005T
5	Thickness of brake facing	.198	.125
6	Spring load-pounds at .750 height	9 lbs.	8 lbs.
7	Fit of bearings on high speed planet gear shaft—diameter	.0005L	.0004T
8	Fit of high speed planet gear on shaft-diameter	.0006L	.0002T
9	Fit of bearings in high speed spider-diameter	.0010L	.0004T
10	Backlash of high speed planet gear with high speed fixed ring gear	.015	.002
11	Backlash of high speed planet gear with high speed movable ring gear	.015	.003
12	Fit of bearings in high speed spider-diameter	.0015L	.0005T
13	Fit of bearing in high speed movable ring gear-diameter	.0015L	.0001T
14	Fit of high speed movable ring gear in low speed spider bushing-		
	large diameter	.006	.002
15	Backlash of low speed planet gear with low speed fixed ring gear and		
	high and low speed movable ring gears	.020	.002
16	Fit of low speed planet gear on shaft-diameter	.0005L	.0004T
17	Fit of bearings in low speed spider-diameter	.0015L	.0005T
18	Fit of bearings on low speed spider shaft-diameter	.0005L	.0004T
19	Fit of seal in housing—diameter	Pres	s fit





Refere nce Number	Description	Max.	Min.
20	Fit of high speed movable ring gear in low speed spider bushing-		
-	small diameter	.005	.001
21	Fit of adapter plate pilot in housing-diameter	.007	.001
22	Fit of bearing in low speed spider—diameter	.0015L	.0005T
23	Fit of bearing in low speed movable ring gear-diameter	.0015L	.0001T
24	Length of spacers—Short	.031	.027
	Length of spacers-Long	1.742	1.738
25	Fit of bearings on drive shaft-diameter	.0005L	.0003T
26	Fit of power gear splines on movable ring gear splines-side clearance	.0080	.0005
27	Fit of contact point insulator in connector insulator—diameter	.040	.005
- 28	When contact is in extended position contact point should extend 1/16		
	inch plus or minus $1/32$ inch past rear face of adapter plate		
29	Fit of pigtail insulator in housing insulator-diameter	.040	.001
30	Spring load—Pounds at .562 height	7 lbs.	6 lbs.
31	Fit of bearing in housing-diameter	.0020L	.0003T
32	Fit of bearing on low speed movable ring gear-diameter	.0020L	.0009T
33	Fit of low speed fixed ring gear in front and rear housings-diameter	.006	.000
34	Fit of high speed fixed ring gear in housing-diameter	.002T	.002L
35	Backlash of drive gear with high speed planet gear	.015	.002
36	Fit of sleeve on speed reducer drive shaft-diameter	T8000.	.0000
37	Fit of bearing on sleeve-diameter	.0005L	.0005T
38	Fit of motor housing in speed reducer housing-diameter	.006	.001
39	Fit of bearing in housing sleeve-diameter	.0010L	.0001T
40	Fit of motor shaft tang in speed reducer drive shaft-side clearance	.006	.000
41	Fit of sleeve on motor shaft-diameter	.0010T	.0000
42	Length of brush	.760	.500
43	Brake gap	.020	.010
44	Fit of bearing in motor end shield-diameter	.0015L	.0002T

LIST OF TOLERANCES-HUB ASSEMBLY

Refere n ce Number	Description	Max.	Min.
1	Fit of pin in blade root and gear-diameter	.0005T	.0000
2	Fit of blade gear shoulder in blade root-diameter	.0050	.0010
3	Permissible wear on power gear and blade gear teeth to be measured by comparison with unworn teeth		
	Permissible wear on power gear teeth	.003	
	Permissible wear on blade gear teeth	.003	
4	Fit of bearing on power gear-diameter	.0020L	.0014T
5	Fit of seal in adapter plate—diameter	Pre	ess fit
6	Fit of bearing in adapter plate-diameter	.0020L	.0009T
7	Face of contact from face of hub	.005	.000
8	Fit of seal in blade nut-diameter	Pre	ess fit
9	Fit of bearings in hub diameter	.0041	.0014
10	Fit of bearings on blade root-diameter	.0037	.0003
11	Area of contact cones in hub		75%
12	Separator diameter No. 50 shaft	6.385	6.281
13	Slip ring diameter No. 50 shaft	6.198	6.094
14	Width of spline spaces in hub No. 50 shaft	.379	.376
15	Runout of slip rings with respect to cone seat-must be turned and polished on hub	full is	in .003 ndicator ding
16	Distance between front and rear cone seat gage—diameter	6.791	6.688
17	Fit of adapter plate pilot in hub-diameter	.006	.001
18	Permissible out of Round of Bearing Bores in Hub	.0020	.0000

20

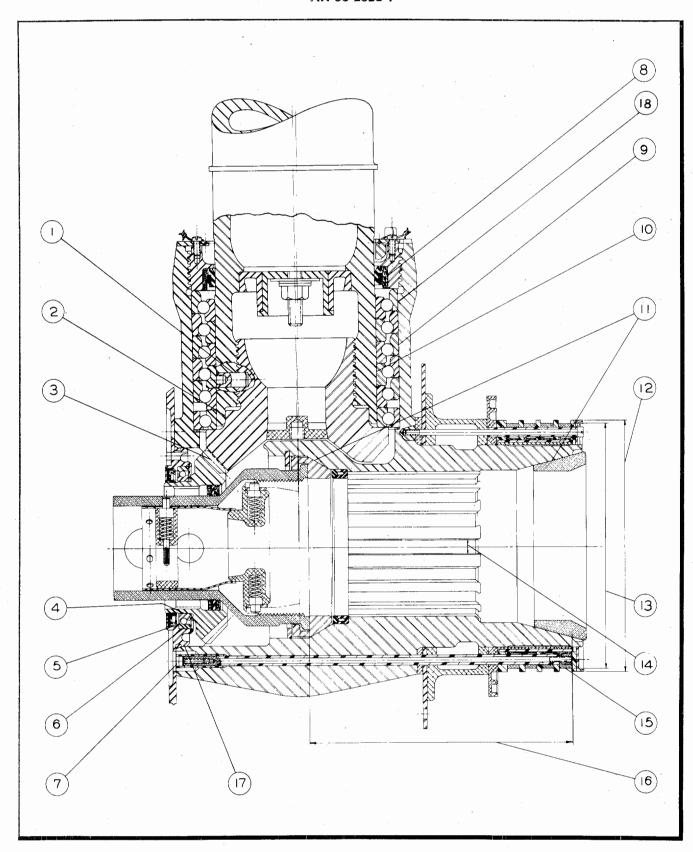


Figure 34—Hub Assembly Tolerances

f. MOTOR AND BRAKE.

- (1) Place electric motor brush assemblies in brush holder.
- (2) Fit the armature into speed reducer high speed drive shaft. Set speed reducer housing on blocks so as to allow the low speed drive gear free movement. Tap end of armature shaft until it is tight in the speed reducer shaft and until the speed reducer gearing is forced as far as it will go toward the low speed end of the housing.
- (3) Slip motor housing over armature, feeding the motor leads through the holes in the front speed reducer housing. Be certain that the motor housing keyway aligns with the key riveted to the front housing.
- (4) Coat motor retaining nut threads with thread lubricant and screw in place with properly fitting wrench. To tighten thoroughly, clamp speed reducer in vise. Apply a torque of 250 foot-pounds then line up the lug and screw holes of motor cover line up with the slot in the nut and screw holes in motor housing respectively. Never back up nut to line up lug and screw holes.
- (5) Install two spacers and shims in place on armature shaft to provide .005 inch to .010 inch between the top of the shims and spacers and the bottom of the bearing seat in motor housing as indicated in "A," figure 42. This is necessary since it eventually controls the end play in the speed reducer assembly.
 - (6) Install bearing in front of motor housing.
- (7) Place solenoid assembly in position and attach leads to terminal posts on motor housing. Make certain that the winding terminals lie flat on their posts, with no tendency to be forced either up or down when the nuts are tight.
- (8) Attach the diaphragm assembly with its coil springs, spacers, and bolts. Thoroughly tighten attaching screws. Locate spacers and shims, insert locating key in armature shaft, and install the brake disc, followed by the plain washer and nut.

- (9) Check brake clearance by using a feeler gage between the brake plate and the inner ring of the sole-noid housing. The clearance should be between .010 inch and .020 inch. Clearance adjustments may be obtained by adding or removing shim laminations on the armature shaft directly behind the front brake disc.
- (10) Install the motor cover. The motor cover must be tried with the locking key in each slot in the ring nut, until one is found which allows the screw holes in the motor cover to line up with the threaded holes in the motor housing. Do not loosen the motor retaining nut to align these holes.

g. LIMIT SWITCHES.

- (1) Install limit switch assemblies, switch arm, and screw terminals in place. Making sure that the limit switch arms do not ride on cam lobes, install gasket, and screw the adapter plate in place. Using a depth gage, determine the distance which the limit switch contacts extend through the adapter plate. The contacts must extend 1/16 inch to 3/32 inch from the face of the plate. When retracted by the cam lobes the contacts must retract a minimum of ½ inch below the face of the adapter plate.
- (2) Attach the motor leads to the limit switch terminals, using spring washers under screw heads.
- (3) Seal the motor terminal connectors in the rear housing with beeswax.

b. POWER GEAR.

- (1) Pack bearing with grease, Specification No. AN-G-4, grade AA.
- (2) Slip the power gear laminated shim on the shoulder of the power gear.
- (3) Press bearing into adapter plate, being careful to have thrust face of outer race against adapter plate and install the bearing snap ring.
- (4) Place grease seal in the adapter plate; press power gear into the bearing, being careful not to damage the seal.

SECTION VII TEST PROCEDURE

1. PRELOAD ADJUSTMENT.

- a. Place the propeller on a checking table and set the blades at the specified low angle.
- b. Make certain that the ring seal is removed from the adapter plate. Mount the power gear assembly on the hub and bolt tightly, using regular power unit attaching bolts with spacers under them.
- c. Remove the bolts and, while holding the power gear assembly firmly in place, measure the clearance between it and the face of the hub. The clearance should be from .002 inch to .005 inch.
- d. If the proper clearance is not had, laminations may be added to, or removed from the power gear shim between the gear and bearing.

2. ADJUSTMENT OF ANGLE RANGE.

If it is desired to raise or lower the low angle setting, proceed in the following manner:

- a. Run the power unit until the low limit switch cuts out.
- b. Remove the power unit from the hub and the power gear assembly from the power unit. Before replacing the power gear assembly, index the desired low blade angle setting with the master spline on the power unit drive.

EXAMPLE

If it is desired to change the low blade angle setting from 20 degrees to 18 degrees, replace the power gear with the 18-degree mark opposite the master mark on the spline. It will be noticed that changing the setting in this manner raises or lowers all blade angle settings alike.

NOTE

A change in low blade angle setting should always be accomplished by changing the indexing of the power gear. However, should a change in any setting other than the low blade angle be desired, an adjustment can be made by filing or welding the cam lobe.

- c. In case a replacement cam is installed, the angles should be carefully checked. Manufacturing tolerances will cause some variation in angular limits even though the replacement cam may have the same part number as the original. It will also be necessary to measure the angular difference between the low angle setting and the point where the mechanical stop becomes effective. This difference may be ascertained by the following method:
 - (1) Mount the propeller on a checking table.
- (2) Making certain that the fixed stop is removed, mount the power unit on the hub.
- (3) Attach a brush assembly to the slip rings. Connect a battery to the brush assembly and energize the "DECREASE RPM" circuit so that the blade angle is increased a few degrees.
- (4) Energize the "INCREASE RPM" circuit until the propeller reaches the electrical low limit.
- (5) Measure the blade angle with a protractor at the 42-inch station of the blade.
- (6) Bypass the low limit switch by introducing a current through the "INCREASE RPM" motor terminal at the rear of the speed reducer housing.
- (7) Operate the motor until the movable stop segment on the bell gear can be seen through the mechanical stop hole in the speed reducer housing.
- (8) Insert one of the fixed stop attaching bolts in the tapped hole of the fixed stop.
- (9) Using the bolt as a finger grip, insert the stop in its hole and ascertain, by touch, if the side of the fixed stop touches the side of the movable stop segment.
- (10) If the fixed stop does not touch the movable segment, remove the stop and operate the motor (as in paragraph(6)), for a few revolutions.

CAUTION

DO NOT OPERATE THE MOTOR WHILE THE FIXED STOP IS IN PLACE.

- (11) Repeat the procedure given in paragraph (9), above, until the fixed and movable stops are touching.
- (12) Measure the blade angle and compare it with the low blade angle. The difference in angles should be 1.6 degrees \pm .4 degree. In the event that the mechanical stop angle is not within the above specified limits, the low pitch cam lobe must be adjusted by filing or welding as required.

3. BALANCING.

- a. Mount the propeller on a balancing mandrel. The blades should be set at an angle approximately in the center of the angle range and held in this position by bolting the power gear assembly in place.
- b. Mount the propeller on an accurate knife edge balance fixture and test balance with each blade in the horizontal and vertical positions. When the propeller is correctly balanced, it will have no tendency to rotate when placed in any position.
- c. If only slightly out of balance, one or more balancing weights may be added to the blade nut locking slots. These weights should be located so that vertical and horizontal balance is obtained.
- d. Should it not be possible to balance the propeller by means of external weights, add putty or modeling clay to the hub barrel of the light blade until the propeller balances correctly. The putty should be placed in the vicinity of the hub barrel index mark so that its position will correspond to that of the balancing material placed on the blade.
 - e. Mark the light blade and weigh the putty.
- f. Place propeller on spindle of a checking table. Remove the light blade.
- g. Remove the gear plug from the gear and the nut and lock washer from the stud. Balancing washers, equal in weight to the putty used in the test balance, should be placed on the stud. Replace lock washers, nut, and gear plug.
- b. Reassemble propeller, replace and lock wire blade nut locking plates, and check for balance as in paragraphs b. and c. above.

4. OPERATIONAL INSPECTION.

- a. Place the propeller on a checking table and set the blades at the specified low angle.
- b. With the fixed stop removed and the low limit switch arm just riding the low limit lobe of the cam, place the power gear assembly on the power unit. Make sure that the specified low blade angle index mark of the power gear is lined up with the indexed spline of the low speed bell gear.
 - c. Mount the power unit to the hub.
- d. Attach a battery of the proper voltage to the slip rings, using a brush assembly strapped to the slip rings. An ammeter must be included in the circuit so that the amperage used in operation can be measured.
- e. Load each blade to 5,400 inch-pounds and complete the "DECREASE RPM" circuit until the ammeter shows a fairly constant reading within the following limits:

Part No. of Motor	Volts	Amperes Drawn
100222	12	20
106800	12	24
106800	24	12
102890	24	10
108951	24	13

NOTE

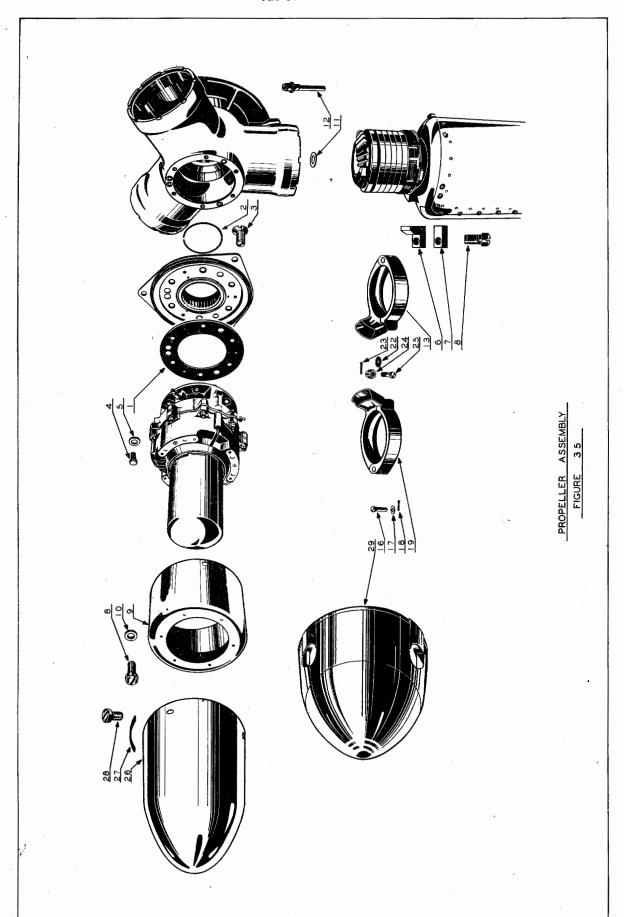
Make sure that the battery is fully charged before taking these readings.

- f. Should more than the specified amperage be drawn, the brake clearance should be inspected as outlined in section VI, paragraph 4.f.(9), and the procedure of paragraph e. above, repeated.
- g. Complete the "INCREASE RPM" circuit until the low blade angle limit switch becomes effective and remove the weights.
- b. Determine that all blade angle settings are properly adjusted. It should be borne in mind that when applying boosted voltage during feather or reverse operation, an overrun of between 2.5 degrees to 3.5 degrees will occur.
 - i. Remove the power unit from the hub.
- j. Make certain that the motor and blade retaining nuts, as well as all screws and bolts, are properly tightened and secured with lock wire, cotter keys, or lock washers as required.

SECTION I

INTRODUCTION

- 1. This catalog is used as the basic catalog covering the Parts List for the Model C5315S-D Curtiss Electric Propellers. The interchangeability of parts or assemblies may be easily determined by referring to the Propeller Group Assembly Parts List, Section II.
- 2. When ordering spare parts, the model of the propeller for which the request is made should be furnished. When ordering power unit assemblies, speed reducer assemblies and/or rear housing assemblies, the complete cam designation must be given. (See page 53.)
- 3. Some of the items included in the illustrations shown may be procured as complete assemblies only. These items are given an index number on the Group Assembly Parts List, Section II, but they are not indexed on the illustration. Parts which are not furnished individually and which must be ordered with complete assemblies have no index numbers. Part numbers are used for identification purposes only.
- 4. The Numerical Parts List, Section III, shows all part numbers used in this catalog and should be used in conjunction with Group Assembly Parts List, Section II, for proper identification of parts. The quantities shown in the Numerical Parts List, Section III, are the total quantities required for installation and are directly indexed to the page number in the Group Assembly Parts List, Section II, on which they may be found. The quantities shown in the Standard Parts List, Section IV, are the total quantities required for installation.
- 5. Symbols and abbreviations used in this catalog are:
 - @ indicates note at bottom of page
 - * indicates a part is not procurable
 - ar indicates quantity "as required"
 - x indicates blade angles at the 42-inch station
 - # indicates parts used on figures 47 and 48
 - ## indicates parts used on figure 47
 - ### indicates parts used on figure 48
 - (†) numbers in parentheses placed next to an assembly or part numbers refer to the series of notes covering "Cuff Assemblies"

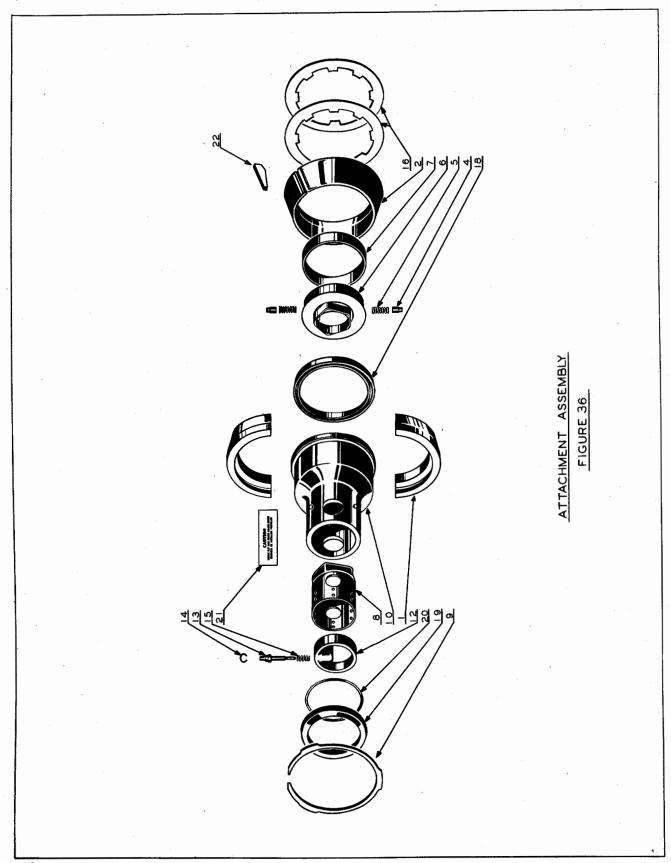


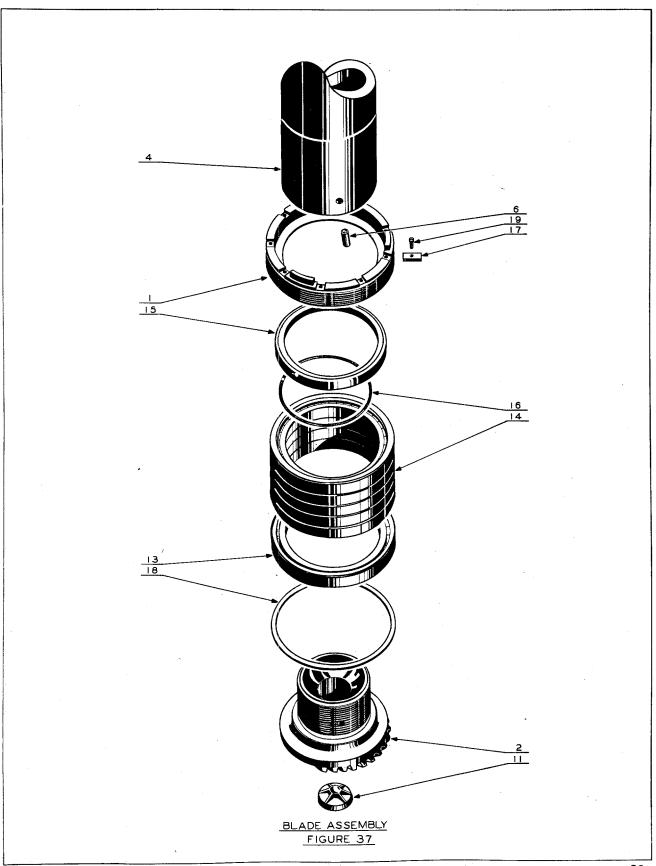
RESTRICTED AN 03-20BE-1 SECTION II—GROUP ASSEMBLY PARTS LISTS

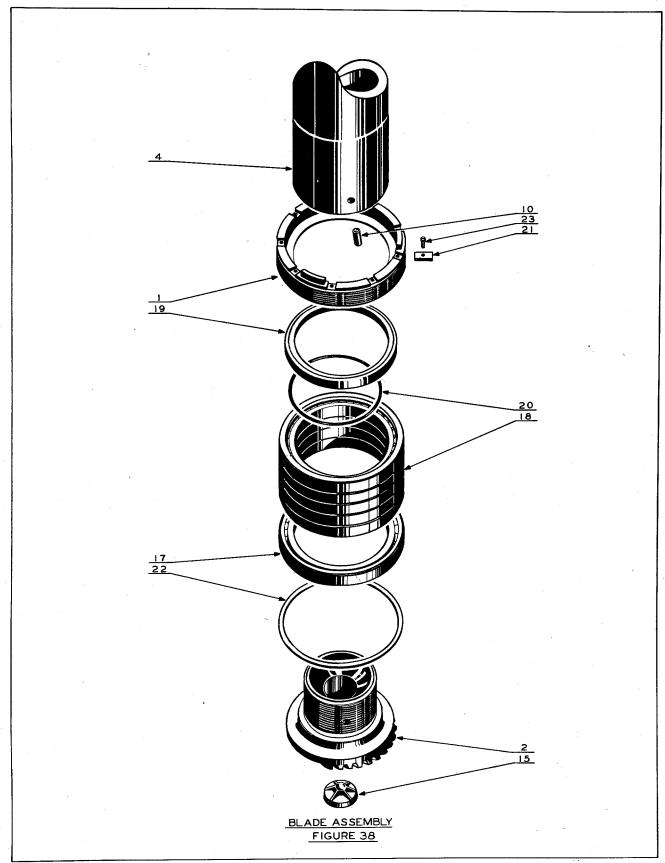
	INDEX NO.	STOCK	GROUP A	GROUP Aircraft Accessories					6	1	UN B	ITS	PER	ASS	1	100	PRO	PROPERTY CLASSIFICATION			
IG.			MAJOR ASSEMBLY Propeller Assemblies						3-D	S-D	S-D	S-D	S-D	S-D	S-D	S-D					
_		E D	PART NUMBER	1	2	3	4	5	6	NOMENCLATURE	0531	0531	C531F	05315	C531F	0531	0531	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
35				Pı		1	ID:	18		sembly - C5315S-D3, D4, 20, D21, D22, D24, D26	l٦	1	1	1	1	1	. 1	1			
35 35	1		101418 101719		Ga	s	ke B	#	1	ower unit er gear retaining	ļ	11366	i	11136	11366	1 3 6 6	- -	11136		03-A 03-A	125M/ 125M/
35 l	1 2 3 4		AN510-10-6		S	$d\mathbf{r}$	dъ	1-	F	lat head	1136	13	3	3	3	3	3	3		29	128/
35 35	4		88126 AN960-616		Bo	1	t he	,	Ţ,o	er unit attaching	6	6	6	6	6	6	5 6	6		03-A 04-A	125M/ 128/
35	5		101728		We	1	gh	ŧ,	+	lain ropeller balance -	1									07.	
35	7		101187		We	ei,	gh	l t	ea -	y ropeller balance –	ar	181	a.	ra:	na:	ra	ra:	rar		03-A	125 M/
35	8		AN502-8-6		S	7	dw	1	1g	it 	ar			na i	۱۵.	na.	na.	rar		03-A 04-A	125 M/ 128/
35 I	9		104057		C	v	er	-	S	llister head eed reducer lain lain				16			16			03-A	125 M /
35 35	10 11		AN960-8 104349		Wa	8	ne he	ļ	1	(lain Plain				6	1	ra:	1	rar		04-A 03-A	128/ 125 M /
35	12		101245		1111	ll h	A+	ДЯ	90	bly - Deicer slinger Assembly - LH propeller								3		03-A	125 M/
25 35	13		106664-1 *106664-3		C	R	je in	g t	qr	assembly - LH propeller ale emale					331111					03-A 03-A	125 M/ 125 M/
35 35 35 35 35 35 35	16		*106664-4 AN394-33			R	in in	\$	+	emale					1					03-A 04-A	125 M / 128/
35 l	17		AN960-416			W	as	цe		·										04-A	128/
35 35	1 <u>8</u> 19		AN380C2-3 106664-2		C					tter Assembly - RH propeller	,				1	3		3	a speci	29 03 - A	128/ 125 M /
35 I	19		*106664-5			R	1n	gł.	+ 1	ale								3		03-A	125 M /
35 35	16		*106664-6 AN394-33		ĺ	R	1n 1n	8	1	emale								1		03-A 04-A	125㎡/ 128/
35	17		AN960-416			W	ឧន	hе							1	3 1 1 1 1 6 3 3 3		11		0)i = A	128/
25 35	18 22		AN380C2-3 AN960-516		W	as	n he	ŋ- '	10	tter Plain					6	6		6		29 04-A	128/ 128/
35	23 24		AN380C2-2 AN310-5		P	1 n	1-	10	dt.	ter tle					6333	3		3331		29 04-A	128/ 128/
35	25 26		AN5-7		В	d 1	ŧ	+	Αi	rcraft					3	3		3.		04-A	128/
35 35 35 35 35 35 35 35 35 35	26 27		106963 108060		C	φv	er	A	.S.S	mbly - Power unit ber covered - single								1		03-A	125 M/
					1		1	Ir	ıl e	e cover			l					3		03-A	125 M/ 128/
35 35 35	28 8		AN502-10-6 AN502-8-6		S	¢r ¢r	ew ew	-	F	llister head llister head				6			6	6		01-A 01-A	128/ 128/
35	29		SP190		S	þi	nn	фr	A	sembly (Air Associates			1	1			-			03-A	
٠.								1	.ne	Bendix, N.J.)			_							05-A	125 M /
36 36			101858-3		A	ţt	ac	hn	ı e n	t Assembly	1	1								03-A	125 m/
36 36	1		104269-3 100951		A	C	ac on	nn e	+ -	t Assembly No. 50 shaft propeller			1				1	1		03-A	125 M /
36	2		100952			1	φn		h	b - front	1	1	1			1	İ	1		03-A	125 M/
36			100806			1			¢r	No. 50 shaft propeller ub - rear Assembly - 3 1/16 in.	1	1	1	1	1	1	1	1		03-A	125 M /
36	4		55931-2				Ι.	1,	1	D. crankshaft Crankshaft adapter	1	1	1	12	12	1	1	1		03-A 03-A	125 M/ 125 M/
			55932				Is	þr	i n	- Crankshaft adapter	2	2	2	2	2	2	2	2		03-A	125 M /
36 l	5678		55930 89859				R	1n	ıak	r – Propeller shaft Crankshaft adapter	1	i	li	1	ļi	1		1 1 1		03-A 03-A	125 M/ 125 M/
36 36 36 36 36 36			89897				oc in	k	Ī	rankshaft nut Tub puller snap ropeller shaft - bronze	1	1	1	111	221111	1	1	1		03-A	125M/
36	.9 10		89146 101795			N	u t	\$ ~	F		1	i .	1	1	4	1		1 1		03-A	125 m /
									-	short	1	1	1	1	1	1	1	1		03-A	125 M/
																				'	

Revised April 5, 1944

RESTRICTED
AN 03-20BE-1



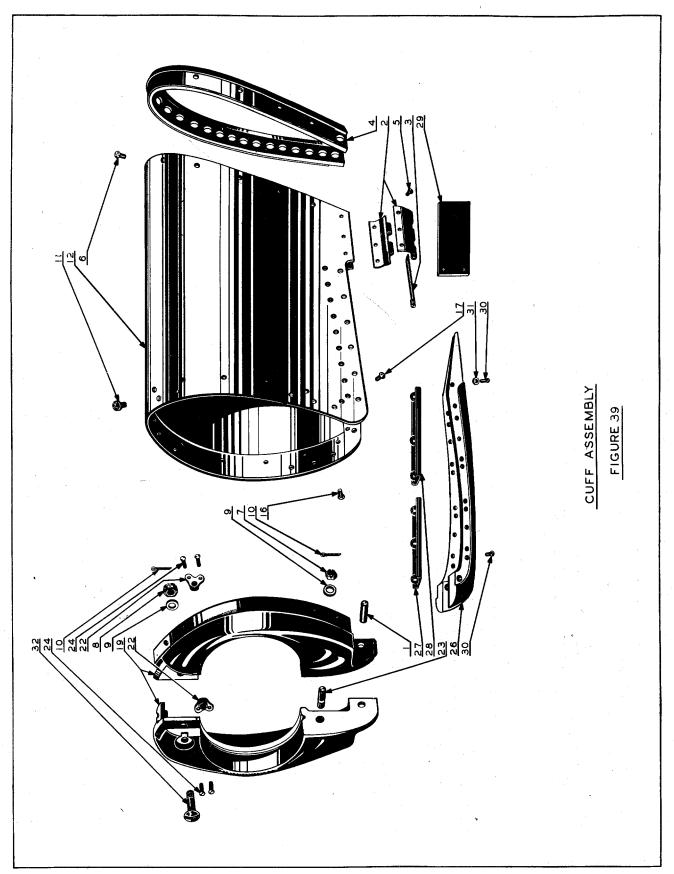




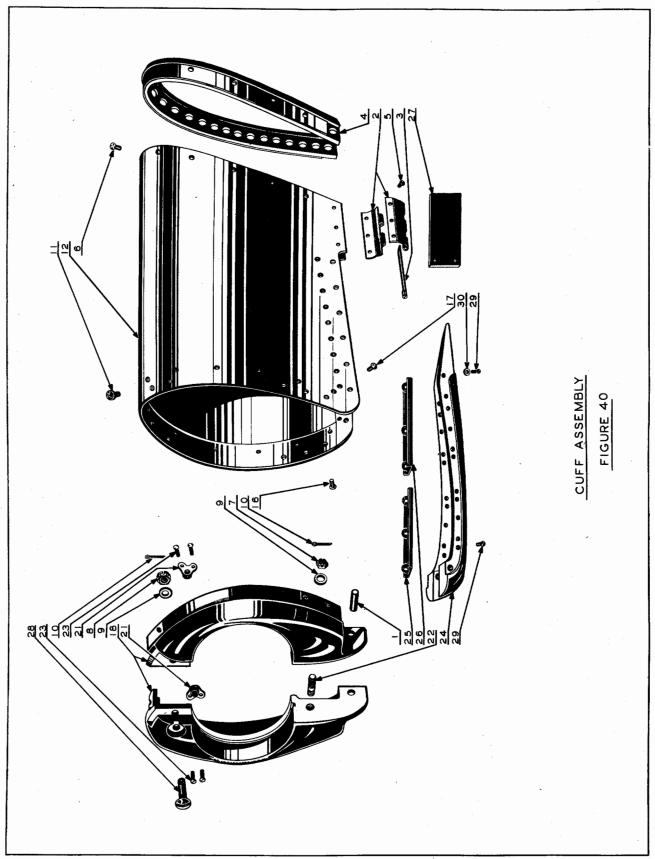
SECTION II—GROUP ASSEMBLY PARTS LISTS

	INDEX NO.	S	GROUP	Aircraft Accessories						F		UNI (XX	TS I	ER A	CU	+	9	PROPERTY CLASSIFICATION					
IG. 10.		C	MAJOR ASSEMBLY	P	Propeller Assemblies									S-D4	S-D1	S-D2	3-D2			3-D2	3-D2	3-D2	
		K E D	PART NUMBER		1	2	3	4		5	6	NOMENCLATURE	C53158-D3	05315	05315	05315	05315	61550	05315	05315	U. S. NAVY	U. S. ARMY	BRITISH
6			100525				I	oc	k	F	ir	Assembly - Crankshaft											•
666	12 13 14		100526 100527 100528					P	1	us n	-	t g - Crankshaft nut lock Crankshaft nut lock ing - Crankshaft nut	1	111	1 1 1	1 1 1	1 1 1	1 1	1 1 1	1 1 1		03-A 03-A 03-A	125M/ 125M/ 125M/
6	15		100871		,				-			lock - Lock pin -	1	1	1	1	1	1	1	1		03 - A	125M/
6	16		89487-2				S	pa	9	er		crankshaft nut Propeller hub rear	1	1				1			•	03 - A	125 M/
6	16		89487-4				S	pa pa	d	er		ne Propeller hub rear		1	1	1						03-A	125M/
6666	18 19 20 21		112287 112323 112324 111668				22	ea	1	ng	I	ne (o. 50 crankshaft haft nut oil ssembly - Tension Brush removal - warning	7	1	1 1	1	1		1 1 1	1		03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/
6	22		101259				Γ	e 8			-	straight lear cone grease	2	1	2	2	2	2	2	2	-	03-A 03-A	125 M/ 125 M/
7			101857-1			В	la	dε				mbly - Interchangeable - LH	3									03-A	125 M /
7			101857-2			B	la	đε	,	Ā٤	8	mbly - Interchangeable		3								03-A	125 M /
7777777777777777777	1 2 4 6 6 6 6 6 6 1 1 1 3 1 4 1 5 1 6 1 6		55737 101811-1 101811-2 102826 101812 615cc1.5-6 614cc1.5-6 101814-1 101814-2 101814-3 101814-4 108355 101490-1 101490-2 108796				M M	Id of the control of	the half of the state of the st	in an	e e g	ade retaining g Assembly - LH blade g Assembly - RH blade Blade - LH Blade - RH - 11 ft 6 in 11 ft 6 in. Blade gear (.3755 in.) Blade gear (.3825 in.) Blade gear (.3990 in.) Blade gear (.4068 in.) Blade gear Assembly - Offset race ig - Preload ig - Active (set of 5) blade retaining nut Blade retaining nut	1	1 1 1 1 ar ar ar 1 1 1 1 1						manyers is a proper description.		03-A 03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/
7 7 7	17 18 19		55927 89144 AN502-8-6				F	Ze 7	,	_	B	al lade nut lock lade preload bearing Fillister head	1 1 1	1 1 1								03-A 03-A 03-A 04-A	125M/ 125M/ 125M/ 128/
8			104928-1			В	le	ıdε				mbly - interchangeable										07	3 OČS4 /
8			104928-2			В	la	de	١	Ās	s	- LH mbly - Interchangeable - RH			7		3	2	2	2		03-A 03-A	125M/
8	1		55737 104927-1				N	Iu t	:	- 1r	B	ede retaining g Assembly - Inter- langeable blade and			1	3	1	1	1	3		03-A 03-A	125 M/ 125 M/
8			104927-2				M	lac	; h	iir	g	ar - LH ng Assembly - Inter-					1					03 - A	125 M/
8	2		104932 104923					0	ie	81 81	g	angeable blade and ar - RH Blade - LH Blade - RH			1	1	1			1		03-A 03-A 03-A	125 M/ 125 M/ 125 M/

Revised April 5, 1944

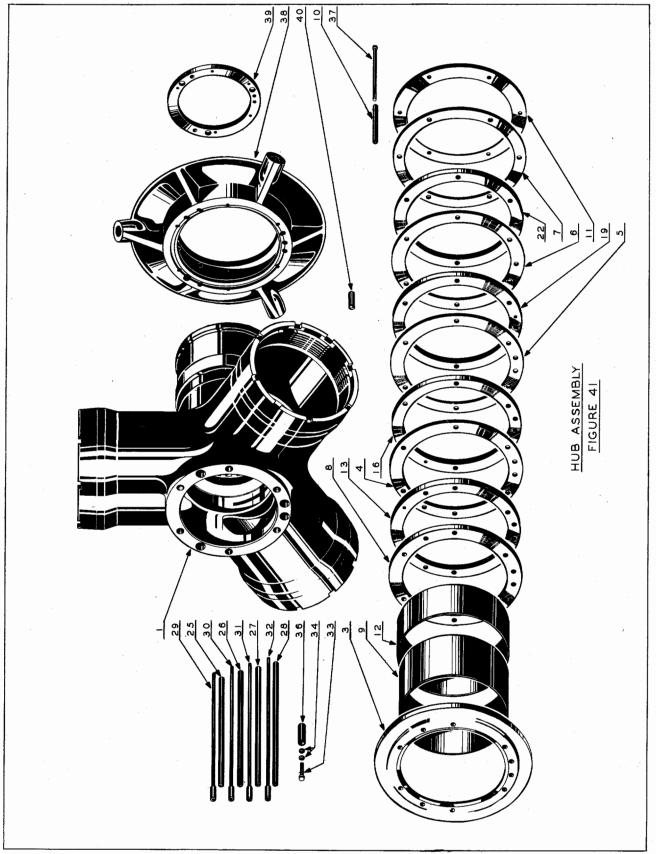


		\$ T	GROUP Air	cra	ft	A	ĊC	es	sories	3	L	UN	ITS	PER	ASS	Y V	* 100	PRO	PERTY CLA	SSIFICATION
FIG. NO.	INDEX NO.	00%	MAJOR ASSEMBLY Pr	ope	11	er	A	នន	emblies	S-D	G-8	1 L	2 E	2 2	<u>u-c</u>	7-2	32-D			
	,	Đ	PART NUMBER	1 ,2	3	4	5	6	NOMENCLATURE	C53158-D3	71530	7577	25.5	277	27.7	1221	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
33888888888888888888888888888888888888	4 10 10 10 10 10 15 17 18 19 20		615-1c1.5-6 614-1c1.5-6 512-1c1.5-9 614-1c1.5-24 512-1c1.5-15 652-1c1.5-6 101814 101814-2 101814-3 101814-4 108355 101490 101490-1 101490-2 108796 108797		S	B.B.B.B.P.P.P.P.P.B.B.B.B.B.B.B.B.B.B.B	a a a a a a a a a a a a a a a a a a a	- - - rg	- (11 ft 6 in.) - (10 ft 3 in.) - (10 ft 0 in.) - (9 ft 9 in.) - (11 ft 6 in.) Blade gear (.3755 in.) Blade gear (.3825 in.) Blade gear (.3911 in.) Blade gear (.4068 in.) Blade gear Assembly - Offset race ag - Preload g - Active (set of 5) Blade retaining nut			a :	l ra: ra: ra:	re	l re re re l	l ira ira ira ira ira	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		03-A 03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/
38 38 38	21 22 23		55927 89144 AN502-8-6		K		-	B B	Blade retaining nut eal ade nut lock slade preload bearing Fillister head			1	. 1		1	1	1 1 1 1 1 1 1 1 1		03 - A 03 - A 03 - A 04 - A	125M/ 125M/ 125M/ 128/
399 39 39 39 39 39 39	12 3 456 78		104344-2 (1) 101739-2 104335-1 104336-1 104185-3 (5) 109901-1 AN425D3-4 AN310-5	C	P: H: P: S: R:	in in in	e - pt	H - cT cA h	ably - Blade inge cuff assembly Trailing edge - blade iff railing edge - blade iff railing edge - blade iff Brazier head 78° angle countersunk sad			111111111111111111111111111111111111111	0				3 1 1 1 6		03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 128/
33333333333333333333333333333333333333	16 17		AN320-5 AN960-516 AN38002-2 525-10-10 104344-8 (2) *104344-5 *104344-5 *104344-6 67304-4 67304-3½ 101767-1 (3)		P: S: SI	in res R R R	he-wtebavv	C-At cett	r - Flat side - Countersunk head - Countersunk head Assembly - Non-deicing			1888	7				1 2 2 2 1 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1		04-A 04-A 29 29 03-A 03-A 03-A 01-C 01-C	128/ 128/ 128/ 125M/ 125M/ 125M/ 125M/ 126HB/ 126HB/
39 39 39 39 39	19 22 23 24	,	104095 *104095-1 *104095-2 101763 101738 AN425D3-7			N	S	po up up	blade cuff rt Detail Assembly port - Camber bort - Flat Elastic stop 5/16 in x 1 11/32 in 78° angle counter- sunk head			1 1 1 2	0				1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/
																		,		



		s	GROUP	Air	c	ra	ft	A	c	ces	sories	_	1	UN a	STIP O	PEI	R AS	SY VI	<u>+ 10</u>		PPO	PERTY CLA	SSIFICATION
iG.	INDEX NO.	TOCK	MAJOR ASSEMBLY	Pr	°0;	pe	11	er	- 1	Ass	emblies	S-D3	四-S	בי	מ מ מ		- N	-2	S-DS				
		Đ	PART NUMBER		1	2	3	4	5	6	NOMENCLATURE	05315	05315	7537	7777		05515	C2212	055158-D24	U. NA	Ş. VY	U. S. ARMY	BRITISH
59 59	25 26 27		104525-2 101700 (4) 101789					St	11	ffε	r Assembly ner - Blade cuff No. 10-32 elastic stop - gang 1 3/4 in.]	L			A THE RESIDENCE OF THE PROPERTY OF THE PROPERT	1			03-A 03-A	125M/ 125M/
9	28		101790					Nu	t	-	spacing No. 10-32 elastic stop			3				- 1	3			03-A	125 M /
9	29		104233-3					Pa	d	-	- gang 2 in. spacing Trailing edge -]					1			03-A	125 M/
9	30		an425d3-6	-				R1	ve	ŧ	stiffener - 78° angle countersunk head			1	1				1			03-A	125 M /
9	31 32		AN960-3 AN25-18				Во	Wa 1t	s)	ner (nead - Plain levis			1	2				1] 2 1			29 04-A 04-A	128/ 128/ 128/
			The nu shown	mbe are	r	o fo	f r	Cu on	f1 e	f A	ssembly components ff assembly only.												*
0			104684-2 (1)		Cu	ff	A	SS no	sen-	bly - Blade - RH - anti-icing				-	,						03´-A	125 M /
0 0	1 2		101739-2 104335-1				Pi Hi	n ng	-	H.1	nge – cuff assembly Trailing edge – blade				3							03-A	125 M /
0	3		104336-1				Ρi	n	_	Cu	ff ailing edge – blade]							03-A	125 M /
0.0	4 5 6		104185-3 (5 109901-1 AN425D3-4)			Ri	ri ve ve	t	As	ff sembly - Chafing Brazier head 78° angle countersunk				1							03-A 03-A 03-A	125M/ 125M/ 125M/
10 10 10 10	7 8 9 10 11		AN310-5 AN320-5 AN960-516 AN380C2-2 525-10-10 104684-8 (2)			Pi Sc	t sh n re	– W	he Ca Sh Co As	ed stle ear Plain tter Washer head sembly - RH - blade				1 2 2 2 2 2					And the second s		29 04-A 04-A 04-A 29 29	128/ 128/ 128/ 128/ 128/ 128/
10 10 10 10 10 10	16 17 18		*104684-4 *104684-5 *104684-6 673D4-4 673D4-3 ¹ / ₂ 101760 (3)					Ri Sp Ri Ri	b ac ve	et et et	ff - Blade cuff Blade cuff - Blade cuff - Countersunk head - Countersunk head Detail Assembly - RH -				1	2 2						03-A 03-A 01-C 01-C 03-A	125M/ 125M/ 125M/ 126HB/ 126HB/
0 10 10 10	21 22 23		*101760-1 *101760-2 101763 101738 AN425D3-7				Nu	Su t	- pi	poi No	n-anti-icing t - Camber - cuff t - Flat - cuff 10-32 elastic stop /16 in. x l 11/32 in. 78° angle countersunk				1	0		the state of the s				03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/
10 10	24 25		104685-2 (4 101789	.)			St Nu	1f t	fe -	ehe	ad r - RH - cuff 10-32 elastic stop				5			-				29 03 - A	128/ 125 M /
0	26		101790				Nu		-	ge No	ng • 10-32 elastic stop -				2			-				03-A	125 M /
0 0 0	27 28 29 30	•	104233-1 AN25-18 AN425-D3-6 AN960-3				Ri	lt ve	t	Ī'1 - C	ng ailing edge - stiffener levis 78° countersunk head Plain				1 2	2				Wilder of the second se		03-A 03-A 04-A 29 04-A	125 M/ 125 M/ 128/ 128/ 128/
			The n shown	umb ar	e	r f	of or	0	u <u>f</u> ne	f	Assembly components uff assembly only.												
																				And the Control of th			

RESTRICTED
AN 03-20BE-1



RESTRICTED

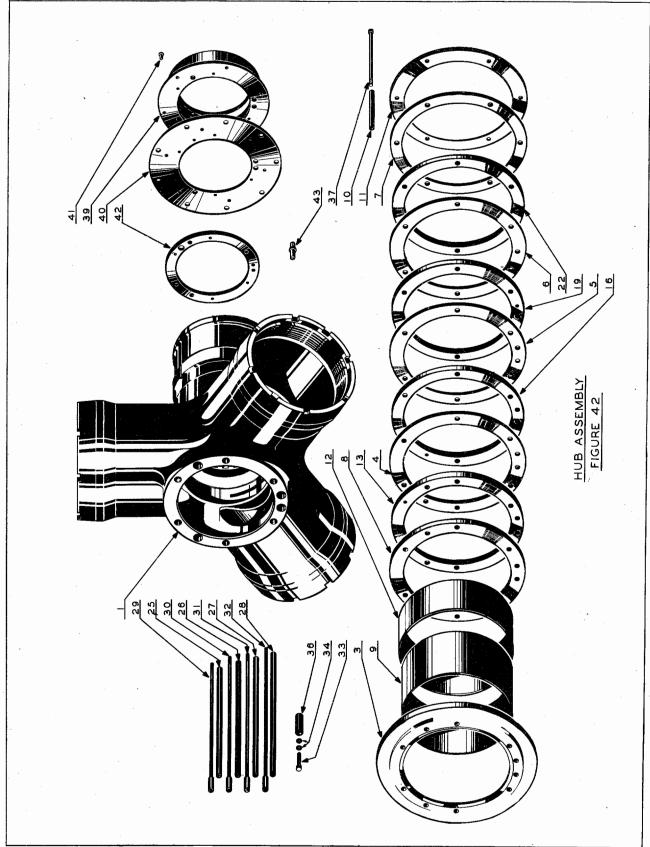
WAR THE BOTH

AN 03-20BE-1

SECTION II—GROUP ASSEMBLY PARTS LISTS

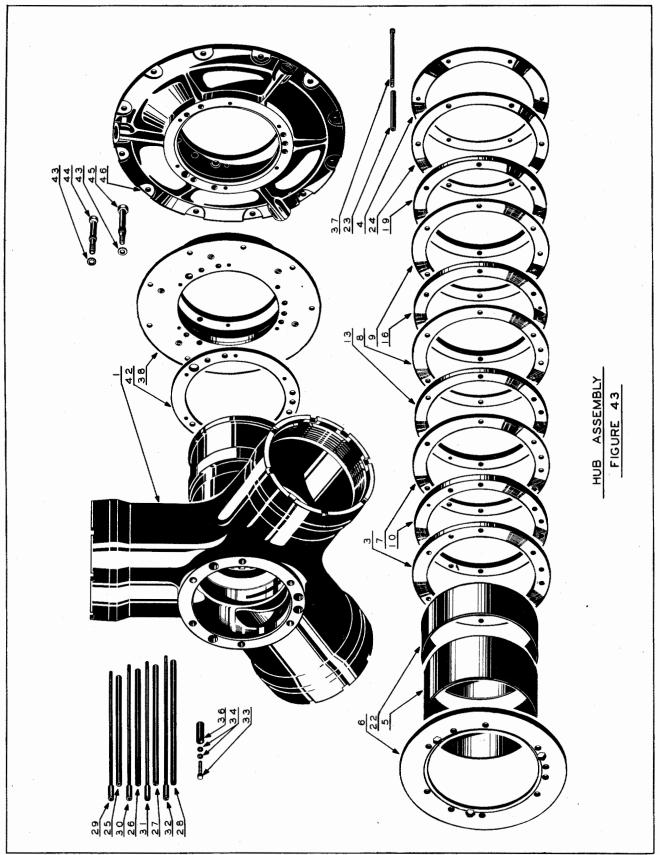
	s	GROUP Airc	·a	f	t.	Αc	cc	e	ssories	F	T	U	מאו	TS	PER	A -	SSY	, И=	+ 1	0	DD C	DEDTY CLA	CCIEICATION
IG. INDEX	\$ 7 0 0 K	MAJOR ASSEMBLY Proj	e e	1:	le	r		As	semblies	3-03	1 5	5	3-D1	3-12	1 5	אח-נ	כח-גי	1 5	2-72	3-D2	PRO	PERIT CLA	SSIFICATION
	ED	PART NUMBER 1	2	3	4	T	5	6	NOMENCLATURE	65315	על און	1077	053158	05315	7 - 7 - 7 - 7	CTCCO	55315		いっつてつ	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
		Cuff Assembly No (1) Cuff Assembly complete unchafing str (2) Sheet Assembly no end of fit trailing education or welded in the street of the except for elastic stc (4) Stiffener Description machined exgang nutrivet holes (5) 104185 Type supplied for existing initially fixerip. When it is cuffs to ne replace chatcheir resembles.	it is the contract of the cont	ydd llitan aiptt harring uibhnec	as -bea Act 11 h fpfic dast	With the said said said said said said said said	ili rini su	l comb	be supplied as a ted with the 104715 y. lied only undrilled th all ribs and cement strips riveted ly - supplied machined screw holes and tholes. mbly - supplied ttachment screw holes, and pad attachment strips the field which with the 104185 type ssary to refit existing the field and/or so on cuffs not fitted lades, the 104715 thick											0			
41 41 41 41 41 41 41 41 41 41		chafing str	I P	b	A bisssilserie	Person in State St	sesing participation of the sesion of the se	mil restriction of the state of	supplied. bly - Anti-icing - no nner flange ropeller ng Assembly Ring - slip ring stor - Front - slip ring e - Slip ring mg - Slip ring blip ring mounting mg - Slip ring Lock Slip ring insulator And Block Assembly ck - Slip ring And Block Assembly ck - Slip ring And Block Assembly											111111111611111111111		03-A 03-A 03-A 03-A 03-A 03-A 03-A 03-A	125 M/ 1 25 M/ 125 M/

RESTRICTED
AN 03-20BE-1

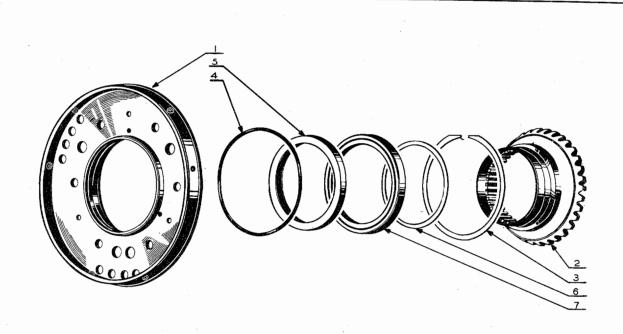


	s	GROUP	Aiı	c	ra	ft	. I	Ac.	ce	ssories	L	I	Ų	NITS	PE	R A	SSY	·	100			
FIG. INDEX	STOCK	MAJOR ASSEMBLY	Pı	0	pe	11	le:	r.	As	semblies	3-13	3-17	1	- DI	שת-נ	3-D2	3-D2	3-D2	3-D2(PRO	PERTY CLA	SSIFICATION
NO. 110.	E D	PART NUMBER			2	3	4	5	6	NOMENCLATURE	05315	05315	7 1 7 1 7 1 7	C15C0	CTCCO	053158	053158	053158	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
41 25 41 26 41 27 41 28 41 29 41 30 41 31 41 33		88606-5 88606-6 88606-7 88606-8 88140-24 88140-25 88140-27 111590-2				Tu Tu Ro Ro Ro	be be d d d d d		H H H	onnector rod insulator connector rod insulator connector rod insulator b connector Power unit cutout									1 1 1 1 1 1 1		03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/
41 34 41 34 41 36 41 37		89107 89107-1 100461 88263-4				Si Si Ti	iir iir		s	itch contact contact point contact point ub contact insulator Special - slip ring									4 ar ar 4		03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/
41 38 41 39 41 40		101215 101282 101714-4				S]	i a	g	a er	taching De-icer slinger Spinner attaching									6 1 1 1		03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/
1234567890123 6 9 2 567890123 4467 4 444222222233333 3533 4467		101880-3 101787 102213 88666 88255-1 88255-2 88255-3 88258 88257 102211 88260 102212 88261 *88256-1 *88256-5 *88256-3 *88256-6 *88259 88256-6 *88259 88256-7 *88256-8 *88256-8 *88259 88256-7 *88259 88256-8 *88259 88256-7 *88259 88256-8 *88259 88256-1 *88259 88256-2 *88259 88256-3 *88256-1 *88259 88256-2 *88259 88256-2 *88259 88256-3 *88256-7 *88259 88256-8 *88259 88606-5 88606-7				HS THIRRER SST	bil SSSSIIISBRTR R R bebeet	s - apppsses in bir RB in RB in RB	PI PI PRESIDENCE OF THE PRESID	k - Slip ring And Block Assembly k - Slip ring Connector rod insulator Connector Co		1116111111111111111 4 2 2 4 4 2 2 2 4 2 2 2 4 2 2 2 4 2 2 2 4 2 2 4 2 2 2 2 4 2									03-A 03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/

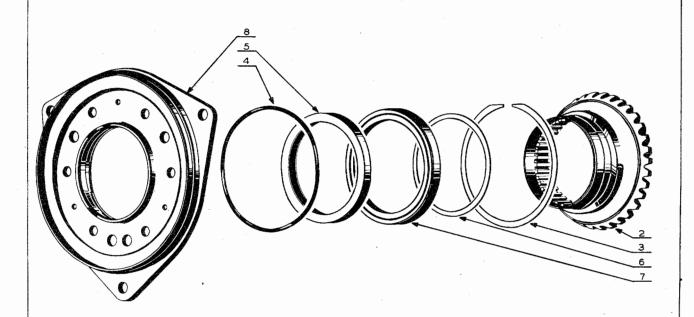
RESTRICTED
AN 03-20BE-1



			GROUP	Air	cr	a f	`t	Α	cc	es	sories	T		UN	ITS	PER	ASS	,	100			
IG.	INDEX	\$ 7 0 C	MAJOR ASSEMBLY						_			-129-	F	ĮĢ.)2U-	20-	-n9	בקת-	1026	PRO	PERTY CLAS	SIFICATION
10.	NO.	K E D	PART NUMBER		op 2	1		1	_	6	embl1es NOMENCLATURE	3158	3158	3158	3158	3158	3158	3158	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
		-	Tract Homes	╨.		1	1	-	<u> </u>	ů	HOMENCEATORE	5	, C	18	S.	/ F	15	, E	15	NOT I	ANTI	
12			100791			9	T S	9.0	er	Δ	ssembly - Slip ring											
12	39		100790		i	ľ	-	-		fl	anged - Slip ring - flanged	1	1					İ			03-A	125 m/
											- spinner	1	1								03-A	125 M/
2	40		100792					- 1			- Slip ring spacer - bolting	16	16								03-A	125 M /
12	41 42		AN430-6-10 100793			S	T De	₹1 3 C	ve er	t -	- Round head Slip ring	6	6								29 03-A	12 8/ 125 M/ .
12	43		101267			I	ò	vе	1	-	Spinner attaching	įī									03-A	125M/
3			101880-6		н	ub		Λs	a e	mh	ly - Spinner flange -											
3			101880-7			1			nc	n	anti-icing ly - Spinner slinger -			1	1			נ		,	03-A	125 M/
	,		•		ľ		١		an	t1	-icing					1				*	03-A	125 M /
3	1 2		101787 104871					ip	F	in	opeller g Assembly			111111111	1	1	. 1	1	.		03-A 03-A	125 M/ 125 M/
3 3	2 3 4		88257 102212					In Ri	su no	la -	tor - Front slip ring			1	1	11	1				03-A 03-A	125 M/ 125 M/
3			102211				Š	31	ee	vε	Lock - Slip ring mounting	ļ		ī	ī	11].	1	. [03-A	125M/
う 3	56 78		104836 88 2 55 - 1	1				Se Se	al Da	ra	Slip ring tor - Slip ring			1	1	11					03-A 03-A	125M/ 125M/
<u> </u>			88255-2		ł		1	Se	pa	ra	tor - Slip ring	į		į	1	li	1	1	.		03-A	125 M /
<i>)</i> 3	9 10		88255-3 88256 - 1				ŀ	3e 31	pa ng	ra A	tor - Slip ring nd Block Assembly			1		1					03-A 03-A	125M/ 125M/
3			*88256-5 *88259	İ					R1	ng				1	111	1	1	1			03-A	125 M/
2 3	13		88 2 56 - 2				I	Ri	nø	A	k - Slip ring nd Block Assembly			i	1	1	1	1			03 - A 03 - A	125M/ 125M/
3 3			*88256-6 *88259							ng	k - Slip ring			1	1	. 1	$\begin{vmatrix} 1 \\ 1 \end{vmatrix}$		-		03-A 03-A	125 M/ 125 M/
333333333333333333333333	16		88256-3	ŀ			i	Ri	ng	A	nd Block Assembly			11111116	1 1 1 1 1 1 1 1 1 1 1 1	i	. 1	li	.		03-A	125M/
ク 3			*88 2 56 - 7 *88 2 59						K1 Bl	ng oc	k - Slip ring			1	11	1					03-A 03-A	125M/ 125M/
3	19		88256-4 *88256-8				•	Ri	ng	A	nd Block Assembly			į	į	i	. 1	1	.		03-A	125 M /
3			*88 2 59						B1		k - Slip ring			1		1]			03-A 03-A	125M/ 125M/
3	22		88261 88260					ľu	be	-	Slip ring insulator g - Slip ring			1	1	16	11	16			03-A	125 M/
3	23 24 25 26		88 2 58	ŀ				Ιn	sı	118	tor - Slip ring				1	1	1	11	. .		03-A 03-A	125 M/ 125 M/
3	26		88606-5 88606-6			1	'u 'u	be be			onnector rod insulator onnector rod insulator			1 1 1 1 1	1	1	. 1.	11	.		03-A 03-A	125 M/ 125 M/
3 3	27 28		88606-7			7	լսի	be	-	1	onnector rod insulator			i	i	ļi	11	1			03-A	125M/
3	29		88606-8 88140-24				[ˈu] ₹de				onnector rod insulator b connector			1	1	1	1	11			03-A 03-A	125 M/ 125 M/
3333	30		88140-25			F	≀da	αl	-	lπι	h connector			i	i	i	i	ļi			03-A	125 M/
3	31 32		88140-26 88140-27			F	≀qa ≀da	a d	_	Hu	b connector b connector			1	1 1 1 1	1	1	1			03-A 03-A	125 M/ 125 M/
3	33		111590-2			Ī	٠d	ir	t	-	Power unit cutout			-		1		1	l			
3	34		89107			2	sh:	im	_	1 (itch contact ontact point						ra:				03-A 03-A	125 M/ 125M/
3333	34 36		89107-1 100461			5	h 'u	in		10	ontact point			bi	h:	rh:	ra:	rla.	n		03-A	125 M/
3	37		88 2 63-4						W	-	ub contact insulator Special - slip ring						4		1		03-A	125 M /
3	38		104887			-	3718	ad	eı	at P	taching ssembly - Spinner			6	6	6	6	6			03-A 03-A	125M/ 125M/
3			104888 104889				-	Sp	ac	er	- Spinner			1	i			1	.		03-A	125 M /
3			104889 AN430-6-10					rı Ri	ar ve	ge t	- Spinner - Round head			16	16			16			03 - A	125 M / 128/
3	42 43		104890			2	spla	ad	er	-	Slip ring spinner			1	1			1			03-A	125 M /
3	44		104835 104758-1			5	sd:	rd	w	_	Spinner attaching			3	3	3	3	3			03-A 03-A	125 M/ 125 M/
333333333	45 46		104758-3			5	₽d:	re	W	-	Spinner attaching			Γ	1	3	13				03-A	125 M /
ン	40		106892			1	1	1 n	ge	r	- Anti-icing					1	1			,	03-A	125 M /

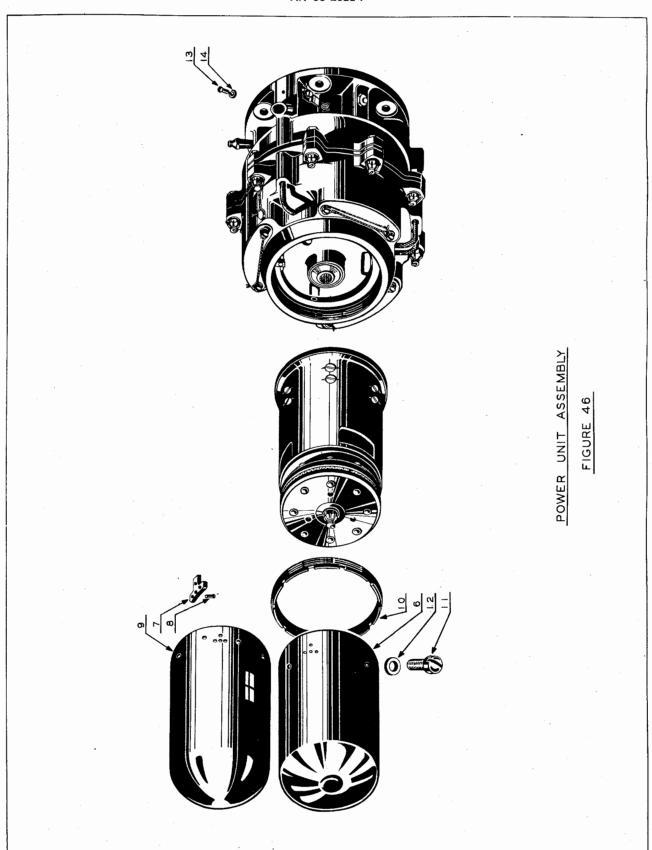


POWER GEAR ASSEMBLY
FIGURE 44

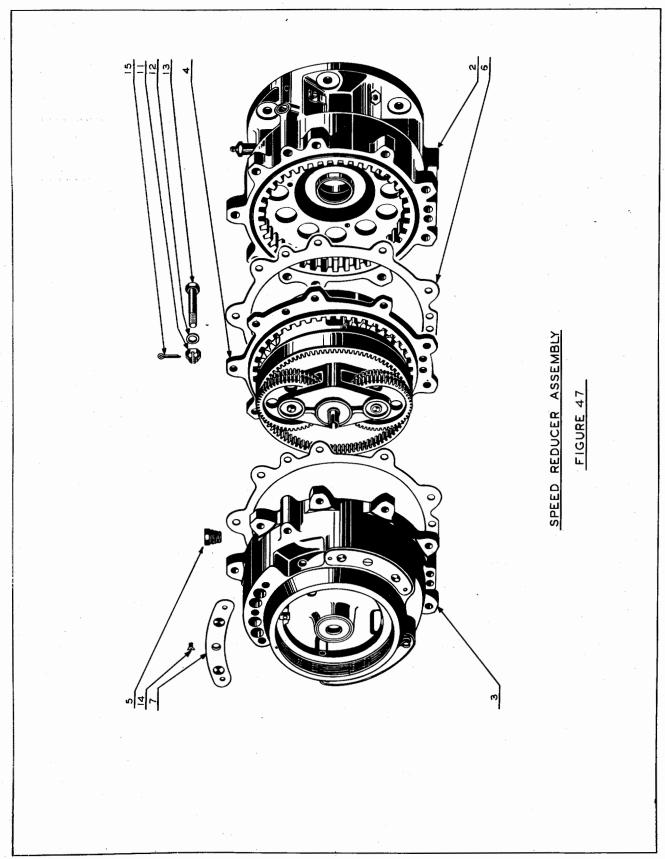


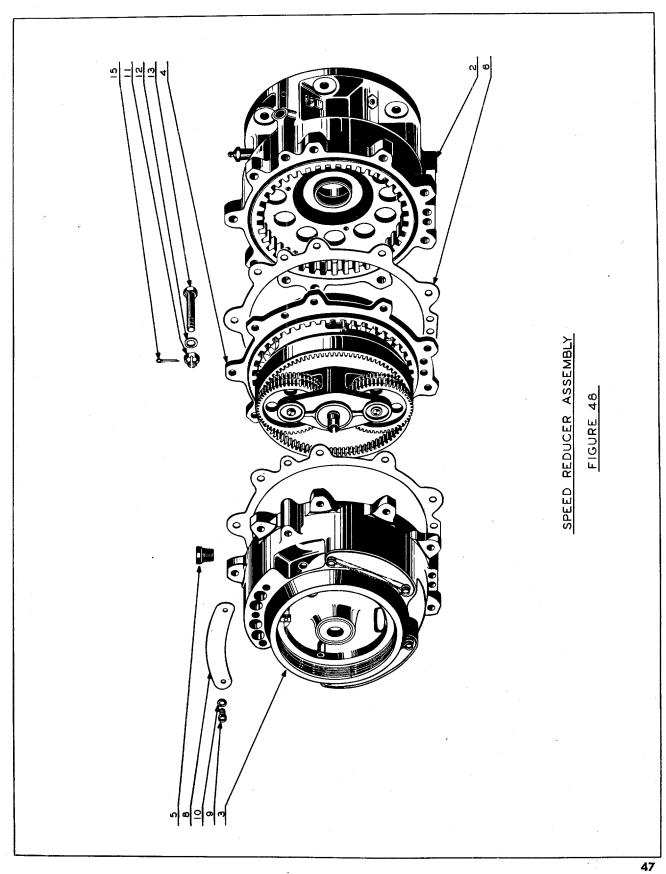
POWER GEAR ASSEMBLY
FIGURE 45

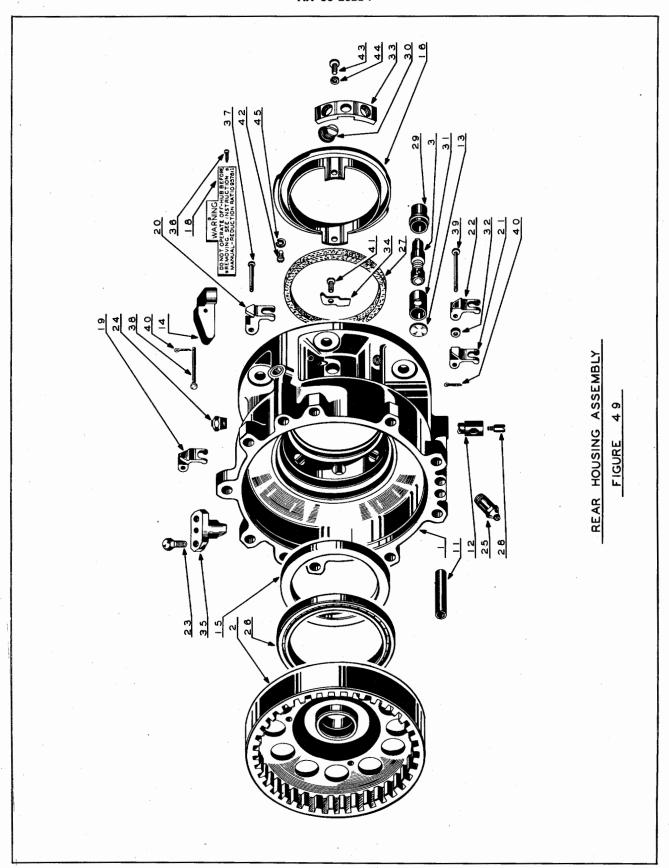
	S	GROUP	Air	c	ra	ft	I	Ac	се	ssories	-	1	ии 8	T\$ I	ER A	SSY CV	7.7	56	PRO	PERTY CLA	SSIFICATION
IG. INDE	ر اي.		Pr	0]	pe	11	eı	r	As	semblies	S-D	S-D	S-D	S-D	S-D	32-D	S-D	S-D			
	D		1	i ,	2	3	4	5	6	NOMENCLATURE	05315	05315	C53158-D18	05315	05315	05315	05315	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
14 14 14 14 14 14 15 16 7	THE THE PERSON NAMED AND ADDRESS OF THE PERSON NAMED AND ADDRE	104266-2 102349 101841-2 101705 101425 100970 55948 55936				Ad Ge Ri Se Se	al al al al	t	er	mbly - Power - Speed reducer Fower spiral bevel Fower gear bearing lock Ring - Adapter grease Speed reducer adapter Fower unit bevel gear - Power unit								1111111		03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/
8 2 3 4 5 6 7 15 15 15 15 15 15 15 15 15 15 15 15 15 1		106794-1 106794-2 106632 101841-1 101841-2 101705 101425 100970 55948 55936	@		ie	Ad Ge Ri Se Se	ar ar ar ar al	st	er -	mbly - Power mbly - Power - Speed reducer Fower spiral bevel Fower spiral bevel Fower gear bearing lock Ring - adapter grease Speed reducer adapter Fower unit bevel gear - Power unit	1 1 1 1 1 1 1 1 1	1 1 1	1	1 1 1 1 1	1	1 1 1 1 1 1 1 1	1 1 1 1 1			03-A 03-A 03-A 03-A 03-A 03-A 03-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/
		indexin	106 g of	٤	ge	ar	•			Assembly except for							,				
16		102600-12		1	- [-		1	2	t Assembly - RH - volt				1	THE THE THE THE THE THE THE THE THE THE		1			03 - A	125 M /
16 16		101773-4			- 1	ĺ			(educer Assembly - RH -				ì			1			03-A	125 m /
16		100222-2							e (A	ssembly - Power unit lectric - 12 volt nef. see page 57) ssembly - Diaphragm ype - 12 volt nef. see page 59)				1		.,	1			03-A	125 M /
16		102600-16					_			t Assembly - RH -										,	
16		101773-8		1	ı	- 1		1	2	volt educer Assembly - RH -			1					1		03-A	125 M/
6		100222-2						l	1(Ref.)			1					1		03 -A	125 m/
6		101375-1							e (A	lectric - 12 volt Ref. see page 57) ssembly - Diaphragm			1					1		03-A	125 m /
										Ref. see page 59)			1					1		03-A	125 M /
.6		104044-7		I	ď	ve:	r	U	ni	t Assembly - LH -						-					`
6		101773-7			- 1	- 1	,	2	4 R	volt sducer Assembly - RH -	1									03-A	125 M/
6		100581-1		-					(A	Ref.) ssembly - Power unit	1									03 - A	125 M/
		101775 0							1(lectric - LH - Ref. see page 55)	1					-				03-A	125 M /
6	1	101375-2		1	ĮI	31/8	a k	qе	A	sembly - Diaphragm ppe - (Ref. see page 59)			1	1						03-A	125 M/



			GROUP Airc	cr	aí	٩t	: 1	λc	c	es	sories	\vdash	1	UN	UTS	PE	R A	SSY	1				
FIG.	INDEX	5 7 0				_	_		_			12	Ġ	اج		-וזקר	- 122	-D22	-D24	-D26	PRO	PERTY CLAS	SIFICATION
10.	NO.	004#0	MAJOR ASSEMBLY Pro	_	_	_	Т	Т	A	88	emblies	158	158	ו ה ה	2 2	122	158	158	158	C53158-D26	U. S.	U. S.	BRITISH
_		-	PART NUMBER 1		2	3	4		5	6	NOMENCLATURE	053	053	1 E	3 6	S	653	653	053	053	NAVY	ARMY	DRITISTI
46			104044-8					,	TT~	١,,	Assembly - RH -												
6			101773-8	1				1	24	1	olt ducer Assembly - RH -		1							0.000		03-A	125 M/
6			100581-2			•	1			(I	ef.) sembly - Power unit		1									03-A	125 M/
			100501-2			171		١	1	e.	ectric - RH - lef. see page 55)		1									03-A	125 M/
6			101375-2			B	ra	k	е	As	sembly - Diaphragm pe - (Ref. see page 59)		1									03-A	125 m/
6			108328-7]	Pa	w	er		Ur Ur	11	Assembly - LH -						1					03 - A	125 M /
6			101773-7			S	pe			Ŗ	ducer Assembly - LH -						1					03-A	125 m /
6			102890-1			M	dt	9	r	A:	sembly - Power unit ectric - LH -												
6			101375-2			В	ra	k	е	(I A	ef. see page 59) sembly - Diaphragm						1					03 -A	125 M /
										t	pe - (Ref. see page 59)						1					03-A	125 M/
1 6			108328-8		Pc	w	er		U Se	11	Assembly - RH -							1				03-A	125 M /
1 6			101773-8			S	pe			R	ducer Assembly - RH -							1				03-A	125 M /
46			102890-2			M	dt	þ	r	A	sembly - Power unit lectric - RH -							-					
46			101375-2			В	ma	lk	e	()	ef. see page 59) sembly - Diaphragm							1				03-A	125 M/
			101919							t	pe - (Ref. see page 59))						1				03-A	125 M/
			The following Power Unit Ass	10	a r mb	t 1	1 ε	e s	r	•	ommon to the preceding												
46			102888			c	dv	re	r	A	sembly - Electric tor - blunt nose	,					1	1				03-A	125 M /
46	6		104660					; 6	v		- Electric motor -		1					ı	1			03-A	125M/
46 46	7 8		100407 100550				I	2	g	- t	Electric motor cover - Electric motor cover	1	ĺ				1					03-A	125 M /
46			89887			C	φī	re	r	A	lug sembly - Electric	4	4				4	4				03-A	125 M /
46	. 9		104662					2 6	v	r	tor and timer			1		1			111	1		03-A 03-A	125M/ 125M/
46 46	7 8		100407 100550				F	2	v	t	Electric motor cover - Electric motor cover			,	 -	ь			1,1	h		03-A 03-A	125 M/
46 46	10		88119 AN502-8-4								lug otor housing retainer Fillister head	1	1			1	1	1	1	13		03-A 04-A	125M/ 128/
46 46 46	12		AN960-8 AN920B5-4			V	γa¦s	зþ	ıe	r	Plain Round head	3	3	1	3	3	3	3	13344	334		04-A 29	128/ 128/
46	13 14		AN935-6			V	74 5	31	16	┝¯	Medium spring lock	4	4	1	<u> </u>	4	4	4	4	4		29	128/
			One pint of Co be supplied wi Assemblies.	ur i t	t. h	2	3 0		3p	ee of	Reducer 011 No. 2 will the above Power Unit												
											•												

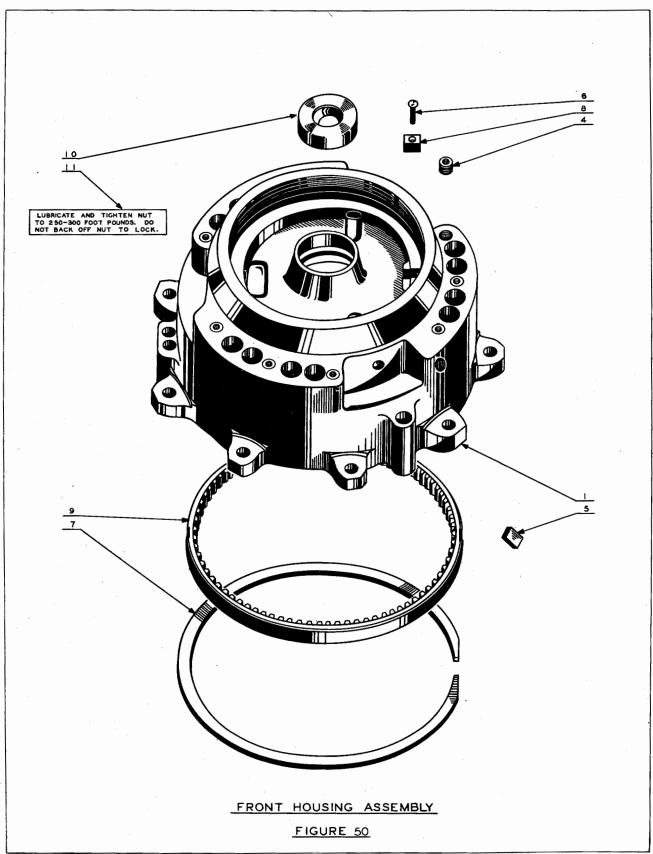






		S	GROUP	Air	°C	ra	ft	I	lc.	ce	ssories	F		INU	T S F	ER A	SSY	4	9	PRO	PERTY CLA	SSIFICATION
FIG.	INDEX NO.	C K	MAJOR ASSEMBLY	Pr	° 0	ре	11	eı	· ·	As	semblies	S-D3	S-D4	S-Dl	S-D2	S-D2	S-D2	S-D2	S-D2			
	.,	E D	PART NUMBER	1		2	3	4	5	6	NOMENCLATURE	05315	05315	05315	05315	05315	05315	05315	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
47			101773-4				ST	96	а	R	ducer Assembly - RH				1			1			03-A	125 M /
47	2		101704-4					Ηc	us	i	g Assembly - Speed reducer - rear				ı			1			03-A	125 M /
47	3		101423				1	Ηc	us	11	g Assembly - Speed reducer - front				1			ı			03-A	125 M /
+7	4		101478				1	Ge	aı	1	ssembly - Speed reducer				1			1			03-A	125 m /
48 48	2		101773-7 101704-3								ducer Assembly - LH g Assembly - Speed	1				1					03-A	125M/
+0 +8	3		101704-3							1	reducer - rear g Assembly - Speed	1				1					03-A	125 M /
48	4		101478							1	reducer - front ssembly - Speed reducer	1				1					03 -A 03 - A	125 M/ 125 M/
18	•		101773-8 101704-4				Sp	ee	d	Re	ducer Assembly - RH		1	1			1		1		03 - A	125M/
18	2										g Assembly - Speed reducer - rear		1	1			1	١,	1		03 -A	125 M /
+8 +8	3		101423 101478				-		ĺ	1	g Assembly - Speed reducer - front ssembly - Speed reducer		1	1			1		1		03-A 03-A	125M/ 125M/
			The followi Speed Reduc	ng	pa A	a r 3 S	ts em	a bl	re	9 8	ommon to the above											735
¥	5		101316	, .				P1	uε	3	Pipe - speed reducer	1	1	1	,	1	1		1		03-A	125 M /
#	6		101417				P	Ga	sl	de 1	- Speed reducer housing	-							2		03-A	125 M /
##	7		101718-1				þ	Re	te	ı i	er - Counterweight -				3			3			03-A	125 M /
###	8.		102504-1				h	Re	ta	i	er - Counterweight -	3	3	3		3	3	[-		03-A	125 M /
### ###	9 10		AN502-8-6 AN960-8								- Fillister head - Plain	66	6	366		366	366	ı	366		04-A 04-A	128/ 128/
### ### # ###	11 12		AN310-5 AN960-516					Νu	t		Castle - Plain	19 18	19 18	9 18	19 12	9 31.8	9 18	9 11.8	9 118		04-A 04-A	128/ 128/
#	13 14		112374 AN510-8-7				ŀ	Bo	1	Н		9	9	9	9	9	9	93	9		03 - A -	125M/ 128/
##	15		AN380C2-2					Pi	n	- W	Cotter	8	8	8	8	8	8	8	8		29	128/
			## These	1tle	ents	з	aph	pe	a	d (n figures 47 and 48 n figure 47 n figure 48											
49			101704-3					Ho	u	11	ig Assembly - Speed	1				1					03-A	12EM /
49			101704-4					Ho	u	i	reducer - rear g Assembly - Speed reducer - rear	1	7	1	1	-	,	,	1		03-A	125M/ 125M/
49	1		101373						н	du.	ing - Speed reducer -	1			l	1		l	1		03-A	125M/
49	2		101443						G	a	- Movable ring - low speed				ļ	l		l	1		03-A	125M/
											2,000	_	-					-)**/

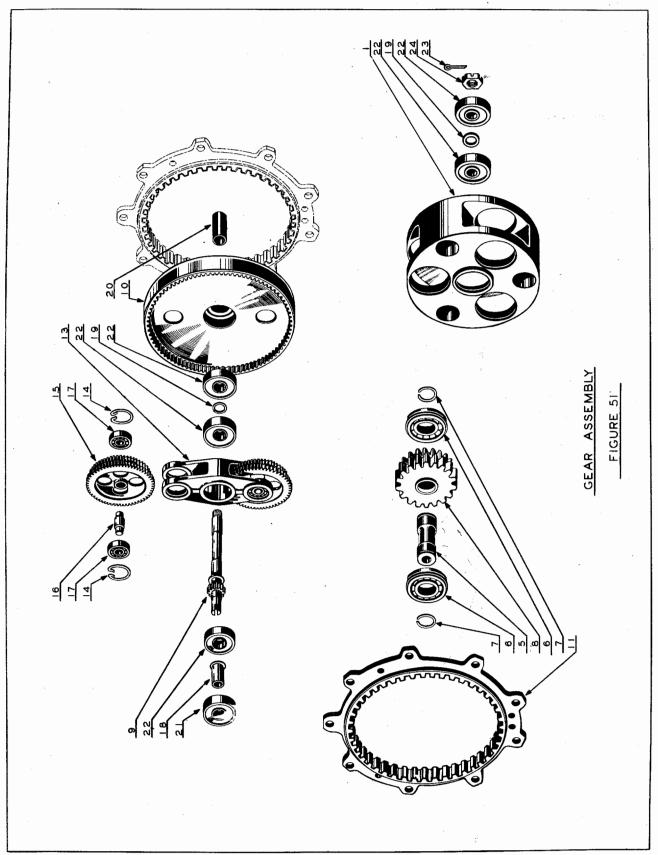
RESTRICTED
AN 03-20BE-1



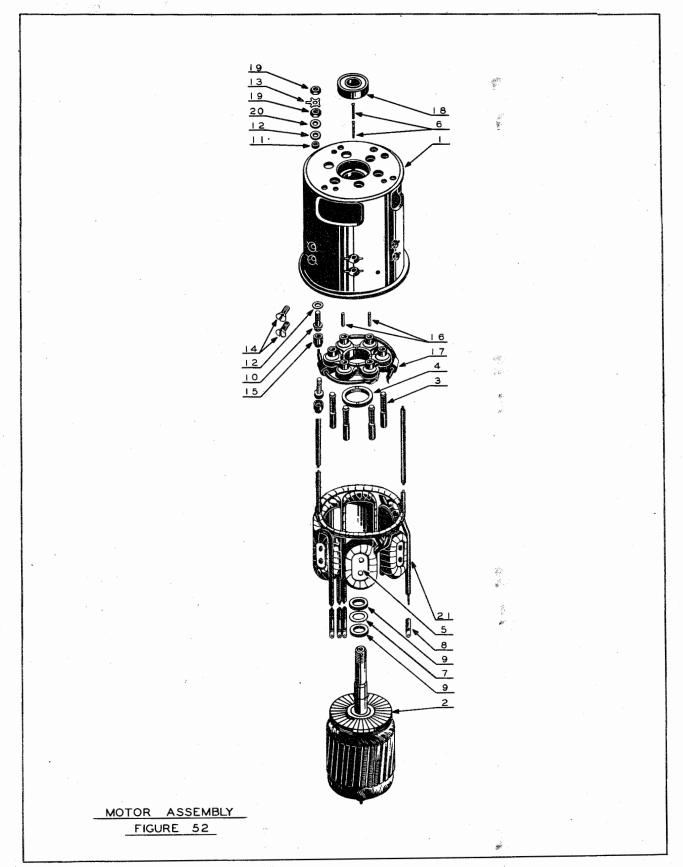
RESTRICTED AN 03-20BE-1

SECTION II-GROUP ASSEMBLY PARTS LISTS

	.,,,,,,,		GROUP	Ali		ra	ft		lc.	ce	ssories		_	UNI	TS I	ER	SSY					
	ILIDEV	S			_							A A	컴	D18	050	D21	D22	D24	026	PRO	PERTY CLA	SSIFICATION
FIG. NO.	NO.	O K	MAJOR ASSEMBLY	Pı	?0	рe	11	er	• ,	la:	semblies	58-	58-	58-	58-	58-	58-	58-	58-	3		,
		Ď	PART NUMBER		1	2	3	4	5	6	NOMENCLATURE	1550	0531	0531	0531	0531	0531	C531	C53158-D26	U. S. NAVY	U. S. ARMY	BRITISH
.																						
49	3		109809						P	1	t Assembly - Power unit cutout switch contact	4	4	4	4	4	4	4	4		03 - A	125 m /
49			*111958-1	-						Po	int - Power unit cutout switch contact		1	1	1	lı	1	1	1		03-A	125 M /
49			*109192							I	sulator - Power unit cutout switch		_	-	-		-	-				
49			*109185							P,	contact point int - Cutout switch	1	1	1	1	1	1	1	1		03 - A	125 M /
49			*102572							1	floating contact ate - Power unit cutout		1	1	1	1	1	1	1		03 - A	125M/
					İ	-					switch arm bearing		1	1	1	1	1	1	1		03-A	125 M /
49			*88141-2								rminal - Power unit connector	1	1	1	1	1	ĺ	1	1		03-A	125 M /
49			*102576				ļ			P;	gtail - Power unit cutout switch	1	1	1	1	1	1	ı	1		03-A	125M/
49			*109084							Sı	ring - Power unit cutout switch	1	1	1	1	1	1	1	1		03-A	125M/
49	11		100268	Ì					I	១៩៤	lator - Power unit motor lead	4	4	4	1	.4		İ	4		03-A	125M/
49	12	and the second	101484		İ				I	sı	lator - Power unit				l'			'				
49	13		100460						D-		stud	4	4	4	4	4	4	4	4		03-A	125M/
			_								e - Insulator - speed reducer connector	4	4	4	4	4	4	4	4		03-A	125 M /
49	14		102351		İ					`	ht - Balance - speed reducer		1	1	1		1	1			03-A	125 M /
49 49	15 16		101420 109876	@						ea. am	- 0il - Power unit cutout	1	1	ī	1		1	1	1		03 - A	125 m /
49			109874-	@					Ca	am	switch - LH - Power unit cutout	1				1					03 - A	125 M /
49	18		101296-1		-				P-	let	switch - RH e - Speed reducer		1	1	1		1	1	1		03 - A	125 M /
49	19		109873							m	warning - Cutout limit switch -	1	1	1	1	1	1	1	1		03 - A	125 M/
49	20		109872								feathering - Cutout limit switch -	1				1					03-A	125 M /
49 49									_	m	feathering		1	1	1		1	1	1		03-A	125 M /
_	21		109871							m	 Cutout limit switch - short 	1	1	1	1	1	1	1	1		03-A	125 M/
49	22		109870							?111	- Cutout limit switch - long	1	1	1	1 2	1	1	1	1		03 - A	125 M /
49 49	23 24		101735 104426		-				Bo F:	oli Liti	- Special ing - Speed reducer				1	1	2	1	2		03-A	125 M /
49	25		102604						ĺ	1	relief icator	1 2 1	1 2	1 2	1 2	1 2	1 2	1 2	1 2		03-A 03-A	125 M/ 125 M/
49 49	26 27		55936 55870						В	a i	ing - Power unit - Speed reducer	1	1	1	1	1	1	1	1 2 1		03-A 03-A	125M/ 1 25M/
49	28		88143	1							- Connector - power unit	1		1		i	4	1	1 1		03-A	125 M /
49	29		88136 ~ 1						Ιı	າຣເ	lator Assembly - cutout	1				ļ.	4	1				
49	30		101744		- Annie State Control of the S				В	11	switch - Mechanical stop						2				03 - A	125M/
49	31		89358		The state of the s				II	sı	segment lator - Cutout switch	۷	2	2	2	2	2	2	2		03 - A	125 M /
,	7.0		000.50		-						pigtail - front terminal	4	4	4	4	4	4	4	4		03-A	125M/
49 49	32 33		89 2 98 89375								er - Cutout switch arm - Low speed ring gear	1			İ	1	1				03-A	125M/
											- speed reducer	1	1	1	1	1	1	1	1		03-A	125 M/
			@ See page cam desi						ac	е	angles and complete											
			53m 4001			- 1																

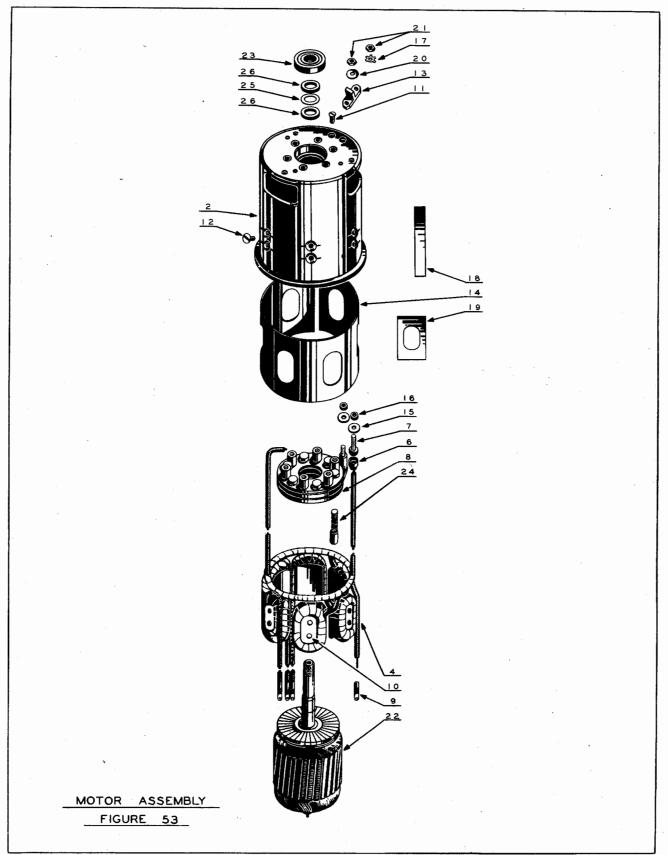


		5	GROUP Aircraft	Accessories	Τ.	0	אוז: ס	S PI	ER A	CU	<u></u>	0	PRO	PERTY CLA	SSIFICATION
IG. 10.	INDEX NO.	OCK	MAJOR ASSEMBLY Propell	er Assemblies	קר קרוני	2 2	S-D	S-D	S-12	SS-D	S-D	SS-D			
		E D	PART NUMBER 1 2 3	4 5 6 NOMENCIATURE	こしてい	52217	0551	05315	05315	05315	C53158-D24	05315	U. S. NAVY	U. S. ARMY	BRITISH
49	34		101840	Plate - Switch arm pin											3 OF 16
49	35		89295	locking Stop - Rear housing - speed		1		1				1		03-A	125 M /
9999999999	3789012345		AN535-0-2 AN592-27 AN392-39 AN392-55 111488-1 500A10-5 AN520-10-5 AN520-8-5 AN935-8L AN935-10L	reducer Screw - Drive - Round head Pin - Flat head Pin - Flat head Pin - Flat head Pin - Cotter Screw - Fillister head Screw - Round head Screw - Round head Screw - Medium spring lock l Washer - Medium spring lock l		3 1 1 1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	3 1 1 2 1 1	13111211111	311121	311121111	31121111	3 1 1 2 1		03-A 29 04-A 04-A 04-A 03-A 29 29 29	125M/ 128/ 128/ 128/ 128/ 125M/ 128/ 128/ 128/
			Following are the c subject model prope	omplete cam designations for				,							
49	16		109876-55.5-164.5	Cam - Power unit cutout switch										03-A	128/
49	16		109874-55.5-162	Cam - Power unit cutout										03-A	128/
49	16		109874-44-203.5	Cam - Power unit cutout		-	.	1			1			03-A	128/
49	16		109876-55.5-163.5	Cam - Power unit cutout			+	-	1					03-A	128/
49	16		109874-55.5-161	Cam - Power unit cutout					-	1				03-A	128/
49	16		109874-40.5-165.5	Cam - Power unit cutout switch						-		1		03-A	128/
			Blade Angles at the Low: High: 22.5° 57.5° 22.3° 46.0° 26.0°	Feather:	x		x	_						03-A	128/
			21.5° 55.5° 21.2° 45.0° 26.7° 46.7°	88.0° 45.0° 88.6°				х	x	x	x	x		03-A 03-A 03-A 03-A	128/ 128/ 128/ 128/
50			101423	Housing Assembly - Speed reducer - front]	1	1	1	1	1	1	1	1		03-A	128/
50	1		106648		.	1	1	ı	1	1	1	1		03 - A	128/
50			*101400	Housing - Speed reducer - I	1	- 1	1	1	i	1	1	İ		03 - A	128/
50		-	*104695		1	1	1	1	1	1	ı	1		03 - A	128/
50	4		100383	Bushing - Screw - front housing	9	9	9	9	9	9	9	9		03 - A	128/
50 50 50	5 6 7	Parameter and Pa	101300 AN420-3-10 101702	Rivet - Countersunk head Ring - High speed fixed	L	1	1	1	1	1]1	1		03-A 29	128/
50	8		88164	Key - Speed reducer high	- 1	1					1	}		03 - A	125M/
50	9		101885	speed fixed ring gear 2 Gear - High speed fixed	- 1	1					1	1		03 - A	125 M /
50	10		106837	Seal - 011 - speed reducer	-	i			1		1	ļ		03-A	125M/
50	11		112845	front housing Decal Front housing		1	1	l	li	1	l	l		03 - A 03 - A	125M/
	I	1	1		1	- 1		i	1			1	1	1	



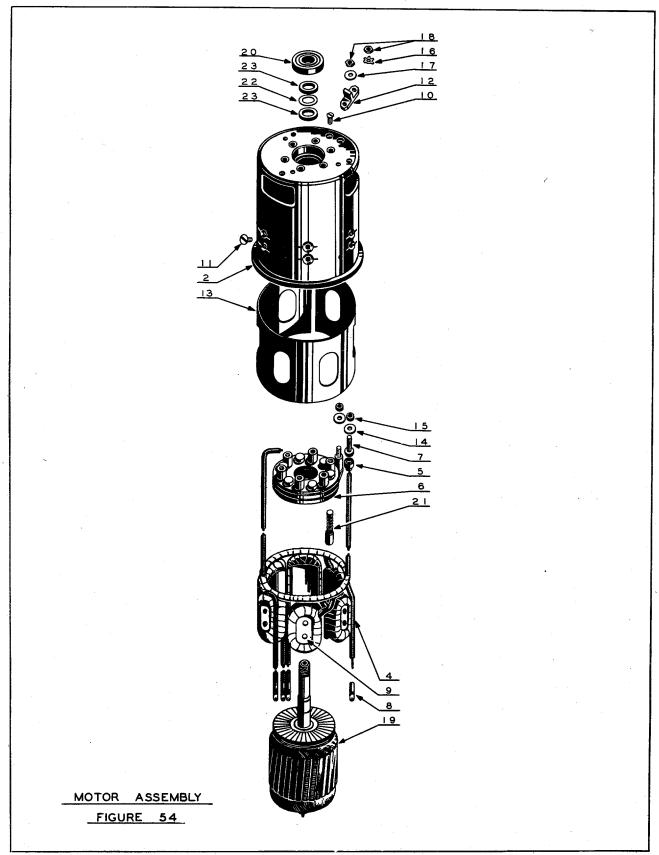
		5	GROUP A	Airo	r	aſ	t	A'c	cc	e's	sories	F		UNI'	rs P	R A	SSY	<u></u>			DEDTY OF	
FIG.	INDEX	T 0 C	MAJOR ASSEMBLY	Pro	.	 1	16	יוב		99	emblies	-D3	古-	-D]	-D2(-D2	-D2	-D2	C53158-D56	PRO	PERTY CLAS	SSIFICATION
NO.	NO.	K	MAJOR ASSEMBLI		T	Ť	_	_	_	1	emp1165	158	158	158	158	158	158	158	158	U. S.	U. S.	BRITISH
		D	PART NUMBER	1	ŀ	2	3	4	5	6	NOMENCLATURE	053	053	c 53	053	053	053	653	c 22	NAVY	ARMY	DKITISTI
_			202/1020					ړ		l ,	and the state of the state of											
51			101478					-		ĺ	ssembly - Speed reducer - drive		1	1	1	1	1	1	1		03-A 03-A	125 M/ 125 M/
51 51	1		101477 101429						Gr 6		Assembly - Low speed ider Assembly - Low	1	l		1			1			03-A	125M/
51	-		*88656								speed Bushing - Low speed -	ı					1				03-A	125M/
51			*108925								speed reducer Spider - Low speed (optional 101427)	1									03-A	125 M /
51			*101428		l		İ				Bushing - Low speed				1		1		1			
51	5		101424							P	spider - small n - Low speed planet							1			03-A	125 M /
51	6		88654							Ве	gear aring - Low speed -	1	ı	!	3			3	3		03-A	125 M /
51	7 8		88629							R:	speed reducer ng - Snap - low speed	6	6	6	6	6	6	6	6631		03-A 03-A	125 M/ 125 M/
51 51	9		101415 101434						Ge	a 1	ar - Low speed planet - Speed reducer drive	í	i	1	1	1	1	1	1		03-A 03-A	125 M/ 125 M/
51	10		101421								- Movable ring - high speed	1	1	1	1	1	1	1	1		03-A	125 M/
51	11		101406								- Speed reducer - low speed fixed ring	1	1	1	1	1	1	1	1		03-A	125 M/
51 51	13 14		101401 101407						Ge	S	Assembly - High speed ider - High speed	1	111	1	1	1	1 1 1	1	1		03-A 03-A	125 M/ 125 M/
51	14		101413							R:	ng - High speed planet gear bearing			1								
51	15		101414							G	retainer ar - High speed planet	2	4 2	4	42	4	4	2	4		03-A 03-A	125 M/ 125 M/
51	16		101416							1	n - High speed planet gear	24	2	2	2	2	2	2	2		03-A	125 M /
51 51	17 18		111262 88169						S :		aring - Ball ve - Speed reducer	4	4			4	1	4.	4		03 - A	125 M /
51	19		1.00401-1						S]	plad	drive gear er - Drive gear bearing er - Drive gear bearing	1 2	12	2	1	2	1 2	1 2	1 2 1	1	03-A 03-A	125 M/ 125 M/
51 51	20 21		100401-3 108133	-					B	eka :	ding - Ball	11	1	1	1	1	21151	1	1		03-A 03-A	125 m/ 125 m/
51 51	22 23		108128 111488-2						B	ele i	ding - Ball	5	151	5	5	5	5	5	151		03-A 03-A	125 M/ 125 M/
51	24		AN320-6				,		N	ut	- Cotter - Shear	ī	ī	ī	ī	ī	ī	ī	ī		014-A	128/
											·.											
52			100581-1							e.	sembly - Power unit - ectric - LH	1									03-A	125 M/
52			100581-2							A: e:	sembly - Power unit - ectric - RH		1								03-A	125 M /
52 52	1 2		88692 100587								g - Electric motor re Assembly - Electric	1	1								03-A	125 M /
52	3 4		88664							sh.	motor - Electric motor	16	16								03-A 03-A	125 m/ 125 M/
52	1		100192					Nu	t	r	Electric motor brush holder clamping	1	,		-						03-A	125 M /
52 52	5		100191 88659					Pc P1		e .	Piece - electric motor Brush holder locking	6	6 2							a Michael and a second	03-A 03-A	125 M/ 125 M/
52	7		88706								Laminated motor and		l								03-A	125 M /
52 52	8		88137 88707								al - Power unit - Magnetic brake	4 2	242								03-A 03-A	125 M/ 125 M/
52 52	10		101177					Ρc	s.	ti .	Brake binding tor - Brake binding	2	2								03-A	125 M /
ے <i>ر</i>			100100							Ţ,	post	2	2								03-A	125 M/
,											·									i maga		
		1	11 5 19bb		1			Ш	L	1		1		1	_		L					 55

RESTRICTED
AN 03-20BE-1

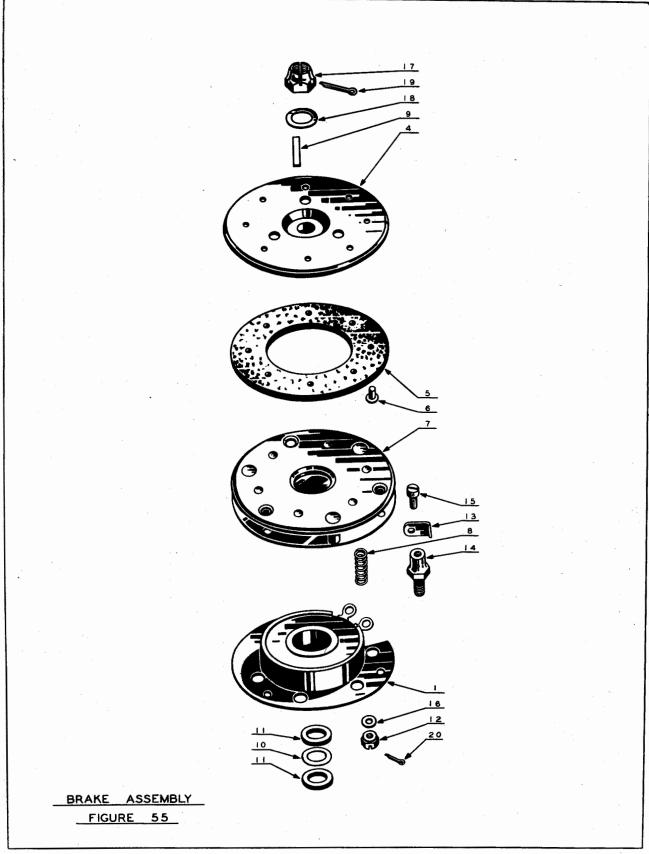


			GROUP Air	·c	re	a f	t		10	ce	9.8	sories			UN	IITS	PER	ASS	Y	-in/	I			
FIG.	INDEX	5 7 0 0					_			_			5	古	-13		152158-D20	5	200	١٤	Ĭ	PROI	PERTY CLAS	SSIFICATION
NO.	NO.	K	MAJOR ASSEMBLY Pr		- T	Т	_	_	1	AS	- 36	emoiles	158	158	158		202	מ ליר	15.0	150		U. S.	U. S.	BRITISH
		D	PART NUMBER	1	12	1	3	4	1	5	6	NOMENCLATURE	053	053	55.3		25.5	7 6	75	3 5	2	NAVY	ARMY	вкизп
555555555555555555555555555555555555555	12 13 14 15 16 17 18 19 20 21		100168 100206 100213 100237 100488 100189 108133 AN345B8 AN960B8 108915-1					WSSPHBNWF		e ld ld sh	er ve er ir er	- Brake binding post - Binding post lock - Pole piece - Binding post Brush holder lock Assembly - Brush g - Ball Hex - Plain Coil Complete Assembly	422211421	4 2 2 2 2 1 1 4	2000								03-A 03-A 03-A 03-A 03-A 03-A 29 04-A 03-A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 125M/ 128/ 128/ 125M/
53			100222-2			1	Мc	t	di		As e1	sembly - Power unit ectric - RH - 12 volt]		1			1	1		03 - A	125 m /
53			109952-2					H	dı	ıs	ir.	g Assembly - 12 volt -]	1	1		İ		1		03 –A	125 M /
53 53 53	2		102889 *109943-2 109938-2						Ī	N1	nd	ing - Electric motor ing Assembly - RH eld Coil Complete			1		1				1		03-A 03-A	125 m/ 125 m/
53			*108971									Assembly - RH Lead Assembly - "P" and			1	-	1		:	1 :	1		03-A	125 m/
53	6		100237									"T" Sleeve - Electric motor binding			1	-	1		1	1 :	1		03-A	125 m/
53	7		101177									post Post - Electric motor			1	.	1		:	1 :	1		03-A	125 M/
53 53 53 53 53 53	8 9 10 11 12 13	-	102891 88137 100191 108711 100213 108733						1 04 04	re Po Bo	rn le re	brake binding lder Assembly - Brush inal - Power unit Piece - Motor brush w - Electric motor w - Pole piece k - Brake terminal			114661		1 4 6 6 2		6	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1466		03 -A 03 -A 03 -A 03 -A 03 -A 03 -A	125M/ 125M/ 125M/ 125M/ 125M/ 125M/
53 53	14 15		108818 100168						ī	In Ja	s u	insulating lator - Electric motor er - Electric motor			3	١	1			1	1		03-A 03-A	125 m/ 125 m/
53	16		100166]	Ιn	ខម	brake binding lator - Electric motor			2	- 1	2		- 1		2		03-A	125 M /
53 53 53 53 53 53	17 18 19 20 21 22		100206 110871 110870 AN960B8 AN345B8 110106					4	N	In In Va Vu	ອນ ອນ ອໄ t	brake binding er - Binding post lock lator - Pole piece lator - Pole piece er - Plain - Hex re Assembly - Electric			4	10001	226624			226624	226624		03-A 03-A 03-A 03-A 04-A 29	125 M/ 125 M/ 125 M/ 125 M/ 128/
53 53	23 24		108133					В	es	ar	ir	motor g - Ball Assembly - Electric			3		1			1	1		03-A 03-A	125 m/ 125 m/
53	25		88706					ı		ı		motor Leminated - motor and			6	5	6		1	5	6		03-A	125 M/
53	26		88707					l				brake - Brake		,	2		2			2 2	2		03 - A 03 - A	125 M/ 125 M/
									The second secon			,												

RESTRICTED
AN 03-20BE-1



			GROUP	Air		ra	ft	: /	\c.	ce	ssories	L	_	ИN	ITS	PER A	ASSY					
FIG.		\$ T O C	MAJOR ASSEMBLY		_		_				gemblies	50-	추	- D18	-D20	-D21	-D22	-D24	-D26	PRO	PERTY CLA	SSIFICATION
NO.	NO.	KED	PART NUMBER		-		3	_	Т	1 6	NOMENCLATURE	53158	53158	53158	53158	53158	53158	55158	053158-D26	U. S.	U. S.	BRITISH
.,				-			Γ		T	†								3				
54			102890-1				M		ļ.,		ssembly - Power unit									·		
_			-				1	1	ı	e	lectric - LH					1					03 - A	125 M/
54			102890-2				M	ł		le	ssembly - Power unit lectric - RH						1				03 - A	125 M /
54 54 54 54 54	}		108715 - 1 108715 - 2					H	du du	ទៅ	ng Assembly - LH ng Assembly - RH					1	1				03-A	125 M /
54	2		102889					-	H	[du	sing - Electric motor					1	11				03-A 03-A 03-A	125M/ 125M/ 125M/
54 54			*108713-1 *108713-2						N N	/1r	ding Assembly - LH ding Assembly - RH					1	1				03-A	125M/
54	4		108915-1						1	F	deld Coil Complete					١,					03-A	125 M /
54			108915-2							F	Assembly ~ LH ield Coil Complete					1) - A	- ,
- 54		Ì		-						1	Assembly - RH						1				03 - A	125 M/
	5		100237								leeve - Electric motor binding post					1	1				03 -A	125 M /
54 54	6		102891							E	older Assembly - Brush ost - Electric motor					1	1				03-A	125 M/
_	7		101177							-	brake binding					1	1				03-A	125 M /
54 54 54 54	8 9		88137 100191								minal - Power unit e Piece - Motor brush	1				1466	146				03 - A 03 - A	125M/ 125M/
54	10		108711						S	sdr	ew - Electric motor						6			İ	03-A	125M/
54 54	11 12		100213 108733						2	di	ew - Pole piece ok - Brake terminal					12	12	2			03-A	125 M /
									1		insulating					1	1				03 -A	125M/
54 54 54	13 14		108818 100168						1	Ins Jas	ulator - Electric motor her - Brake binding					1 1 2	1 2				03 - A 03 - A	125M/ 125M/
54	15		100166]	Ens	ulator - Electric motor											- ,
54	16		100206						1	Ja s	brake ther - Binding post lock					2	12				03-A 04-A	125 M /. 128/
54	17		AN960B8						1.3	via s	her - Plain					2	2				29	128/
54 54	18 19		AN345B8 100587					A			: - Hex ture Assembly - Electric					4	4				03-A	125 M /
			7	ì					1		motor		ĺ			1	1				03 - A	125 M /
54 54	20 21		108133 106980					B	ea In	15) 15)	ng - Ball Assembly					6	Įį 6				03 – A 03 – A	125M/ 125M/
54	22		88706					S	h:	in	+ Laminated - motor and			1								- /
54	23		88707					S	De	a d e	brake r - Brake					2	2				03-A 03-A	125 M/ 1 25 M /
		ļ	101375 1				L		1		ssembly - Diaphragm type											
55			101375-1				ı	l	1	-	12 volt	1		1	1			1	1		0.3 - A	125M/
55			101375-2				В	ra	kε) A	ssembly - Diaphragm type 24 volt		1			,	1				03 - A	125 M /
55	1	-	108665-1					H	φı	ıs:	ng Assembly - Magnetic			ı		1	-		1			
55			108665-2					Н	φι	ıs	ng Assembly - Magnetic	1	1	1	1	1					03-A	125M/
55			102157					D	1 5	3 d	Assembly - Movable	i	i	1	1	i	1	1	1		03-A 03-A	125 M/ 125 M/
55 55 55 55	4 5		101886						I) 1 :	s¢ - Movable ting - Magnetic brake	1	1	1	1	1	1	1	1 1 8		03-A 03-A	125M/ 125M/
55 55	5 6		AN445D4-6						I	3	et - Brazier head	8	8	8	8	8	8	8	8		03-A 29	128/
			·								·						İ					
																				,	1 1	÷*
											·											
		1			_	Ц,	乚	L	1	Ĺ	l			1.		1_	1	L_	1	L		



			GROUP	Atr	ייד ה	e fi	t	A C	ces	ssories			UNI	S P	R A	SSY	41 1				
G.	INDEX	\$ 1 0								Tomb 1 og	-03	古	-D18	-020	-D21	C53158-D22	-D24	Ä	PRO	PERTY CLA	SSIFICATION
Ю.	NO.	OCK E	MAJOR ASSEMBLY		T	_	L e	r .	ASS	semblies	158	158	158.	158	158	158.	158	158	U. S.	U. S.	BRITISH
		D	PART NUMBER	1	2	3	4	5	6	NOMENCLATURE	C53158-D3	053	C53	053	053	053	053	522	NAVY	ARMY	BKIII2H
55	7		101754				D	1 a	ph:	nagm Assembly - Magnetic							-	1			,
55	8		101376				S	pr	in	brake g - Magnetic brake	6	1 6 1	6	1 6 1	6	1	1 6 1	6		03-A 03-A	125 M/ 125 M/
5	9 10		101093 88706				S	нi Hi	-	Magnetic brake shaft Laminated - motor and brake]			1		- 1	1		03-A	125 M /
55	11 12		88707 AN310-4				S	pa	de:	r - Magnetic brake Castle	223	2 2	2 2 3	2	2	2 2 3	2 2 3	2 2 3		03-A 03-A 04-A	125 M/ 125 M/ 128/
5	13		110986				W	as	he	r - Brake diaphragm hold down bolt			l			H		- 1		04-A	125 M /
55	14 15		109092 109093				S	ra cr	de:	r - Brake diaphragm - Brake diaphragm	33331113	333311	333311	33331	33331	3 3 3 3 1 1	33331	33331		03-A 03-A	125 M / 125 M /
555555555555555555555555555555555555555	16 17		AN960-416L AN310-7				W	as ut	he:	r - Plain Castle	3	3	3	3	3	3	3	3		04-A 04-A	128/
55	18 19		AN960-716 111488-5				W.	als	he -	r - Plain	1	1	113	1	113	1	1	1		04 - A 03 - A	128/
5	20		AN380C2-1				P	in	-	Cotter	3	3	3	3	3	3	3	3		29	125 M / 128/
										d*											
															- :						
		ŀ																			
									1	,											
		-																			

RESTRICTED AN 03-20BE-1

SECTION III—NUMERICAL PARTS LIST

				TO			NTIT		110	ł	ļ	<u></u>				QUA		<u> </u>	
PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	053158-D18	C53158-D20	C53158-D21	C53158-D22	C53158-D24	053158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	053158-D18	C53158-D20	053158-D21	C53158-D22	C53158-D24	053158-D26
and the second of		-	_	J	Ŭ	1 7		Ŭ	Ĭ				0	0	8	0	0	0	0
SP190	27			1						100793	41	1	1						
100166	55,57,59	2	2	2	2	2	2	2	2	100806	27	1	1	1	1	1	1	ī	1
100168	57,59	4	4	2	2	2	2	2	2	100871	31	1	1	1	1	1	1	1	1
100189	57	1	1							100951	27	1	1	1	1	1	1	1	1
100191	55,57,59	6	6	6	6	6	6	6	6	100952	27	1	1	1	1	1	1	1	1
100192	55	1	1							100970	43	1	1	1	1	1	1	1	1
100206	57,59	2	2	2	2	2	2	2	2	101037	59	1	1	1	1	1	1	1	1
100213	57,59	12	12	12	12	12	12	12	12	101093	61	1	1	1	1	1	1	1	1
100222-2	43,57			1	1			1	1	101177	55,57,59	2	2	1	1	1	1	1	1
100237	57,59	2	2	1	1	. 1	1	1	1	101187	27	ar	ar						
100268	51	4	4	4	4	4	4	4	4	101215	39								1
100383	53	9	9	9	9	9	9	9	9	101245	27					3	3		3
100401-1	55	2	2	2	2	2	2	2	2	101259	31	1	1						
100401-3	55	1	1	1	1	1	1	1	1	101267	41	ı	1						
100407	45			1	1			1	1	101282	39								1
100460	51	4	4	4	4	4	4	4	4	101296-1	51	1	1	1	1	1	1	1	1
100461	39,41	4	4	4	4	4	4	4	4	101300	53	1	1	1	1	1	1	1	1
100470	45	1	1			1	1			101316	49	1	1	1	1	1	1	1	1
100488	57	2	2							101373	49	1	1	1	1	1	1	1	1.
100525	31	1	1	1	1	1	1	1	1	101375-1	43,59			1	1			1	1
100526	31	1	1	1	1	1	1	1	1	101375-2	43,45,59	1	1			1	1		
100527	.31	1	1	1	1	1	1	1	1	101376	61	6	6	6	6	6	6	6	6
100528	31	1.	1	1	1	1	1	1	1	*101400	53	1	1	1	1	1	1	1	1
100550	45	4	4	4	4	4	4	4	4	101401	55	1	1	1	1	1	1	1	1
100581-1	43,55	1								101406	55	1	1	ı	1	1	1	1	1
100581-2	45,55		1							101407	55	1	1.	1	1	1	1	1	1
100587	55,59	1	1			1	1			101413	55	4	4	4	4	4	4	4	4
100790	41	1	,1							101414	55	2	2	2	2	2	2	2	2
100791	41	1	1							101415	55	3	3	3	3	3	3	3	3
100792	41	1	1			′				101416	55	2	2	2	2	2	- 1	2	2

RESTRICTED AN 03-20BE-I SECTION III—NUMERICAL PARTS LIST

Γ				T	OTA	L G	AUAN	ATITY							TOT	AL C				
	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	053158~D20	C53158-D21	053158-D22	C53158-D24	C53158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	053158-D18	C53158-D20	C53158-D21	053158-D22	¢53158-D24	C53158-D26
	101417	49	2	2	2	2	2	2	2	2	101754	61	1	1	1	1	1	1	1	1
	101418	27	1	1	1	1	1	1	1	1	101760	35				3				
	101420	51	1	1	1	1	1	1	1	1	*101760-1	35				3				
and the second	101421	55	1	1	1	1	1	1	1	1	*101760-2	35				3				
The state of the s	101423	49,53	1	1	1	1	1	1	1	1	101763	33,35			30	30			30	
- Albahania	101424	55	3	3	3	3	3	3	3	3	101767-1	33			3.				3	
PER PER PER PER PER PER PER PER PER PER	101425	43	1	1	1	1	1	1	1	1	101773-4	43,49				1			1	.
on the second second	*101428	55	1	1	1	1	1	1	1	1	101773-7	43,45,49	1				1			
-	101429	55	1	1	1	1	1	1	1	1	101773-8	43,45,49		1	1.			1		1
	101434	55	1	1	1	1	1	1	1	1	101787	37,39,41	1	1	1	1	1	1	1	1
	101443	49	1	1	1	1	1	1	1	1	101789	35			9	6			9	
	101477	55	1	1	1	1	1	1	1	1	101790	35			3	6			3	
	101478	49,55	1	1	1	1	1	1	1	1	101795	27	1	1	.1	1	1	1	1	1
-	101484	51	4	4	4	4	4	4	4	4	101811-1	31	3					,		
	101490	31,33	3	3	3	3	3	3	3	3	101811-2	31		3						
	101490-1	31,33	3	3	3	3	3	3	3	3	101812	31		3		ļ.				
-	101490-2	31,33	3	3	3	3	3	3	3	3	101814	31,33	3	3	3	3	3	3	3	3
	101700	35			3				3		101814-1	31,33	ar	ar	ar	ar	ar	ar	ar	ar
	101702	53	1	1	1	1	1	1	1	1	101814-2	31,33	ar	ar	ar	ar	ar	ar	ar	ar
	101704-3	49	1				1				101814-3	31,33	ar	ar	ar	ar	ar	ar	ar	ar
	101704-4	49		1	1	1		1	1	1	101814-4	31,33	ar	ar	ar	ar	ar	ar	ar	ar
The second second	101705	43	1	1	1	1	1	1	1	1	101840	53	1	1	1	1	1	1	1	ı
	101714-4	39								1	101841-1	43	1				1			
-	101718-1	49				3			3		101841-2	43		1	1	1		1	1	1
	101719	27	1	1	1	1	1	1	1	1	101857-1	31	3							
	101728	27	ar	ar	ar	ar	ar	ar	ar	ar	101857 - 2	31		3						
	101735	. 51	2	2	2	2	2	2	2	2	101858-3	27	1	1						
	101738	33,35	Ï		3	3			3		101880-2	37							,	1
	101739-2	33,35			3	3			3		101880-3	39	1	1						
	101744	51	2	2	2	2	2	2	2	2	101880-6	41			1	1			1	
						1.	1								1					

RESTRICTED AN 03-20BE-1

SECTION III-NUMERICAL PARTS LIST

		L	1	TOTA		AUG	TIT							TOT		AUE	NTIT		
PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	C53158-D22	C53158-D24	C53158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	C53158-D22	C53158-D24	c53158-D26
101880-7	41					. 1	1			104266-2	43								1
101885	53	1	1	1	1	1	1	1	1	104269-3	27			1	1	1	1	1	-
101886	59	1	1	1	1	1	1	1	1	104335-1	33,35			3	3	1	1	3	-
102157	59	1	1	1	1	1	1	1	1	104336-1	33,35	,		3	3			3	
102211	37,39,41	1	1	1	1	1	1	1	1	104344-2	33			3				3	
102212	37,39,41	1	1	1	1	1	1	1	1	*104344-5	33			3				3	
102213	37,39	1	1						1	*104344-6	33			6	-			6	
102349	43								1	*104344-7	33			3			,	3	
102351	51		1	1	1		1	1	1	104344-8	33 **			3				3	
102504-1	49	3	3	3		3	3		3	104349	27					ar	ar	ar	ar
*102572	51	1	1	1	1	1	1	1	1	104426	51	1	1	1	1	1	1	1	1
*102576	51	1	1.	1	1	1	1	1	1	104525-2	35			3	-	_	_	3	-
102600-12	43				1			1		104660	45	1	1			1	1		
102600-16	43			1					1	104662	45			1	1	_	_	1	1
102604	51	2	2	2	2	2	2	2	2	104684-2	35				3				
102826	31	3								*104684-4	35				3				
102888	45	1	1			1	1			*104684-5	35				3				
102890-1	45,59					1				*104684-6	3 5 💡			-	6				
102890-2	45,59						1			104684-8	35				3				
102889	57,59			1	1	1	1	1	1	104685-2	35				. 3				
102891	57,59			1	1	1	1	1	1	*104695	53	1	1	1	1	1	1	1	1
104044-7	43	1								104758-1	41			3	3			3	
104044-8	45		1							104758-3	41					3	3		
104057	27				1			1		104835	41			3	3	3	13	3	
104095	33			3				3		104836	41			1	1	1	1	1	
*104095-1	33			3				3		104871	41			1	1	1	1	1	
*104095-2	33			3		-		3		104887	41			1	1			1	
104185-3	33,35			3	3			3		104888	41			1	1			1	
104233-1	35				3					104889	41			1	1			1	
104233-3	35			3				3		104890	41 🚁			1	1			1	
()															- A - A -	oni	1 5	1	

64

RESTRICTED AN 03-20BE-I SECTION III—NUMERICAL PARTS LIST

					L C	AUG	NTIT							TOT		QUA			
PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	C53158-D22	¢53158-D24	053158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	c53158-D22	C53158-D24	C53158-D26
104923	31			3	3		3	3	3	108715-1	59					1			
104927-1	31					3				108715-2	59						1		
104927-2	31			3	3		3	3	3	108733	57,59			1	1	1	1	1	1
104928-1	31					3				108796	31,33	3	3	3	3	3	3	3	3
104928-2	31			3	3		3	3	3	108797	31,33	3	3	3	3	3	3	3	3
104932						3				108818	57,59			1	1	1	1	1	1
106632	43	1	1	1	1	1	1	1		108915-1	57,59	1				1			
106648	53	1	1	1	1	1	1	1	1	108915-2	57,59		1				1		
106664-1	27				;	3				*108925	55	1	1	1	1	1	- 1	1	1
106664-2	27						3		3	*108971	57			1	1			1	1
*106664-3	27					1				*109084	51	1	1	1	1	1	1	1	1
*106664-4	27					1				109092	61	3	3	3	3	3	-3	3	3
*106664-5	27						1		1	109093	61	3	3	3	3	3	3	3	3
*106664-6	27						1		ļ	*109185	51	1	1	1	1	1	1	1	1
106794-1	43	1				1				*109192	51	1	1	1	1	1	1	1	1
106794-2	43		1	1	1		1	1		109809	51	4	4:	-4	4	4	4	4	4
106837	53	1	1	1	1	1	1	1	1	109870	51	1	1	1.	1	1	1	1	1
106892	41					1	1			109871	51	1	1	1	1	1	1	1	1
106963	27								1	109872	51		1	1	1		1	1	1
106980	57,59			6	6	6	6	6	6	109873	51	1				1			
108060	27								3	109874	51		1	1	1		1	1	1
108128	55	5	5	5	5	5	5	5	5	109874-40.5									
108133	55,57,59	1	1	1	1	1	1	1	1	-165.5	53								1
108328-7	45					1				109874-44									
108328-8	45						1			-203.5	53			1	1			1	
108355	31,33	3	3	3	3	3	3	3	3	109874-55.5									
108665-1	59			1	1			1	1	- 161	53						1		
108665-2	59	1	1			1	1			109874-55.5		ŀ							
108711	57,59			6	6	6	6	6	6	- 162	53		1						
*108713-1	59					1	1			109876	51	1				1			
			l				L								L_	·	Ĺ		نــــــا

RESTRICTED AN 03-20BE-I SECTION III—NUMERICAL PARTS LIST

				TOTA		AUE	NTIT	Y							AL	QUA	NTIT	Y	
PART NUMBER	GROUP LIST PAGE NUMBERS	053158-D3	C53158-D4	c53158-D18	C53158-D20	C53158-D21	053158-D22	C53158-D24	C53158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	053158-D20	C53158-D21	C53158-D22	C53158-D24	354158-no6
109876-55.5								-		55.007	71 77	_	_		_			_	
-163.5	53					,				55927	31,33	3	3	3	3	3	3.		1.
109876-55.5						1				55930	27 27	1	1	1	1,	1	1	1	
-164.5	53	1								55931-2	27	2	2	2	2	2	2	2	
109901-1	33,35	-		18	18			18		559 32 559 3 6	43,51	2	2	2	2	2	2	2	
109938-2	57			1	1	in the second		1	1	559 1 8	43	1	1	1	1	1	1	1	
*109943-2	57			1	1			1	1	614-cc1.5-6	31	1	3	1	+	1	1	1	
109952-2	57			1	1			1	1	615-cc1.5-6	31	3							
110106	57			1	1			1	1	614-1c1.5-6	33						3		
110870	57			6	6			6	6	614-1c1.5-24	33				3				
110871	57			6	6			6	6	615-1c1.5-6	33					3			
110986	61	3	3	3	3	3	3	3	3	652-1c1.5-6	33								
111262	55	4	4	4	4	4		4	4	88119	45	1	1	1	1	1	1	1	
111488-1	53	2	2	2	2	2		2	2	88126	27	6	6	6	6	6	6	6	
111488-2	55	1	1	1	1		1	1	1	88136-1	51	4	4	4	4	4	4	4	
111488-5	61	1	1	1	1	1	1	1	1	88137	55,57,59	4	4	4	4	4	4	4	
111590-2	39,41	4	4	4	4	4	l .	4	4	88140-24	39,41	1	1	1	1	1	1	1	
111668	31	2	2	2	2		2	2	2	88140-25	39,41	1	1	1	1	1	1	1	
*111958-1	51	1	1	1	1		1	1	1	88140-26	39,41	1	1	1	1	1	1	1	
112287	31		-	1	1	1	1	1	1	88140-27	39,41	1	1	1	1	1	1	1	
112323	31	1	1	1	1		1	1	1	*88141-2	51	1	1	1	1	1	1	1	
112324	31	1	1	1	1		1	1	1	88143	51	4	4	4	4	4	4	4	
112374	49	9	9	9	9		9	9	9	88164	53	2	2	2	2	2	2	2	
112845	53	1	1	1	1		1	1	1	88169	55	1	1	1	1	1	1	1	
512-1c1.5			-	_	-	_		-	-	88255-1	37,39,41	1	1	1	1	1	1	1	
- 9	33			3						88255-2	37,39,41	1	1	1	1	1	1	1.	
512 -1c1. 5										88255-3	37,39,41	1	į	1	1	1	1	1	
- 15	33							3		88256-1	37,39,41	1	1	1	1.	1	1	1	
55737	31	3	3	3	3	3	3		3	88256-2	37,39,41	1	1	1	1.	1	1	1	
55870	51	1	1	1	1	1	1		1	88256-3	37,39,41	1	1	1	1	1	1	1	

RESTRICTED

AN 03-20BE-1

SECTION III—NUMERICAL PARTS LIST

			1	OTA			NTIT							TOT		AUE			
PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	C53158-D22	C53158-D24	C53158-D26	PART NUMBER	GROUP LIST PAGE NUMBERS	C53158-D3	C53158-D4	C53158-D18	053158-D20	C53158-D21	d53158-D22	C53158-D24	C53158-D26
88256-4	37,39,41	1	1	1	1	1	1	1	1	89358	51	4	4	4	4	4	4	4	4
*88256-5	37,39,41	1	1	1	1	1	1	1	1	89375	51	1	1	1	1	1	1	1	1
*88256-6	37,39,41	1	1	1	1	1	1	1	1	89487-2	31	1	1	1	1	1	1	1	1
*88256-7	37	1	1	1	1	1	1	1	1	89487-4	31	1	1	1	1	1	1	1	1
*88 2 56-8	37,39,41	1	1	1	1	1	1	1	1	89859	27	1	1	1	1	1	1	1	1
88257	37,39,41	1	1	1	1	1	1	1	1	89887	45			1	1			1	1
88258	37,39,41	1	1	1	1	1	1	1	1	89897	27	1	1	1	1	1	1	1	1
*88259	37,39,41	4	4	4	4	4	4	4	4										
88260	37,39,41	6	6	6	6	6	6	6	6										
88261	37,39,41	1	1	1	1	1	1	1	1										
88263-4	39,41	6	6	6	6	6	6	6	6										
88606-5	39,41	1	1	1	1	,1	1	1	1									!	
88606-6	39,41	1	1	1	1	1	1	1	1										
88606-7	39,41	1	1	1	1	1	1	1	1										
886.06-8	39,41	1	1	1	1	1	1	1	1										
88629	55	6	6	6	6	6	6	6	6										
88654	55	6	6	6	6	6	6	6	6										
* 88656	55	1	1	1	1	1	1	1	1										
88659	55	2	2,													4.			
88664	55	6	6		,														
88666	37,39	1	1					,	1										
88692	55	1	1								-								
88706	55,57,59,61	4	4	4	4	4	4	4	4										
88707	55,57,59,61	4	4	4	4	4	4	4	4			-							
89107	39,41	ar	ar	ar	ar	ar	ar	ar	ar										
89107-1	39,41	ar	ar	ar	ar	ar	ar	ar	ar										
89144	31,33	3	3	3	3	3	3	3	3										
89146	27	1	1	1	1	1	1	1	1						:				
89295	53	1	1	1	1	1	1	1	1										
89298	51	1	1	1	1	1	1	1	1										

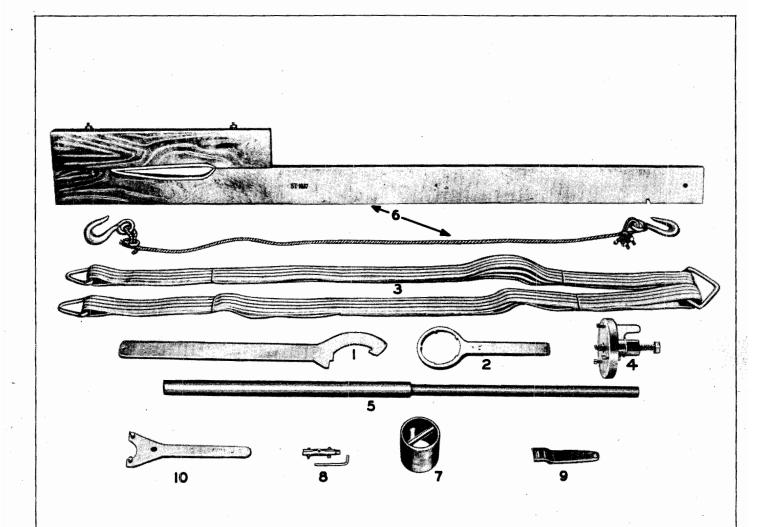
RESTRICTED AN 03-20BE-1 SECTION IV—STANDARD PARTS LIST

				TO	ΓAL	QUAN	ITITY				,	ATTAC	HING	QU/	ANTITY	·	
PART NUMBER	NOMENCLATURE .	C53158-D3	C53158-D4	C53158-D18	C53158-D20	C53158-D21	053158-D22	C53158-D24	C53158-D26	C53158-D3	053158-D4	C53158-D18	C53158-D20	C53158-D21	C53158-D22	C53158-D24	C53158-D26
AN25-18 AN310-4 AN310-5 AN310-7 AN320-5 AN320-6 AN345B8 AN380C2-1 AN380C2-2 AN380C2-3 AN392-39 AN392-39 AN392-35 AN425D3-4 AN425D3-6 AN425D3-7 AN425D3-6 AN425D3-7 AN425D3-6 AN502-8-6 AN502-8-6 AN502-8-6 AN510-8-7 AN510-10-6 AN510-8-7 AN520-8-5 AN520-10-5 AN535-0-2 AN935-8L AN935-8L AN935-8L AN935-10L AN960-3 AN960-8 AN960-416L AN960-416 AN960-716 500A10-5 525-10-10 673D4-3½ 673D4-4	Bolt - Clevis Nut - Castle Nut - Castle Nut - Shear Nut - Shear Nut - Shear Nut - Hex Pin - Cotter Pin - Cotter Pin - Cotter Pin - FH Pin - FH Pin - FH Pin - FH Rivet - CSK hd Rivet - CSK hd Rivet - CSK hd Rivet - Brazier head Bolt - Air craft Screw - Fil hd Screw - Fil hd Screw - Fil hd Screw - FH Screw - RH Screw - RH Screw - RH Screw - RH Screw - RH Screw - Plain Washer - Lock Washer - Lock Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - Plain Washer - CSK hd Rivet - CSK hd Rivet - CSK hd Rivet - CSK hd	8 39 34113411 293	391 1438 111 1 6 8 39 34113411 293 18611	332131434 111 133366 8 39 341134116293 24611 661136	332131434 111 166606 8 39 3341134116293 2611 6653	321 143111111 8 339 34113411 29314611	321 1431111111 8 339 34113411 293124611	3312131434 111 103366 8 39 3341134116293 24611 61136	321 143111111 8 3396 34113411 29312611	9 1 8 3 6 3	36 3	3 12 31 8 36 3 18	3 12 3 1 8 3 6 3 18		12 1 8 3 3 6 3 18		12 1 8 336 3 18

RESTRICTED

AN 03-20BE-1

SECTION V SERVICE TOOLS



Service Tools (C5315S-D)

1 2	ST 1002 ST 1011	Wrench - Blade nut spanner Wrench - Motor nut
3	ST 1012	Sling - Propeller hoisting
4	ST 1015	Puller - Diaphragm brake
5	ST 1034	Bar - Propeller shaft nut
6	ST 1037	Lever - Blade torque testing
7	ST 1038	Protector - Propeller shaft thread
8	ST 1046	Wrench - Hub contact and insulator
9	ST 1077	Wrench - Blade nut spread
10	ST 1073	Holder – Diaphragm brake disk